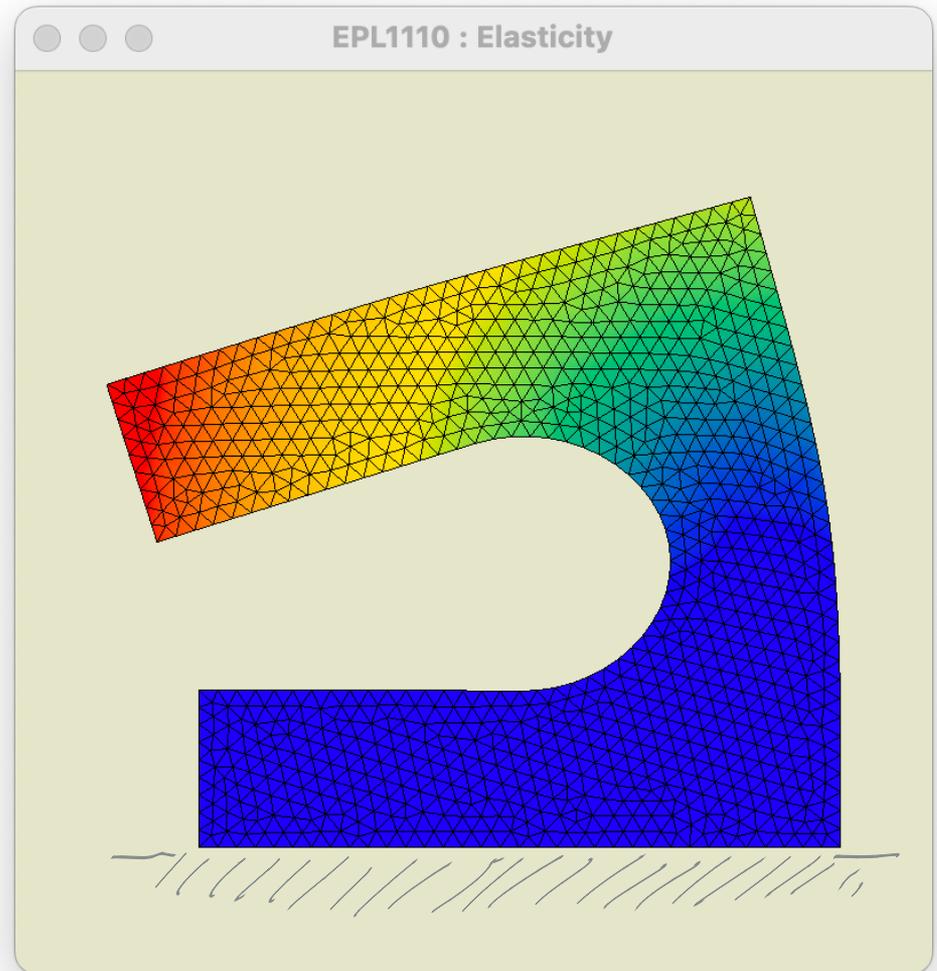
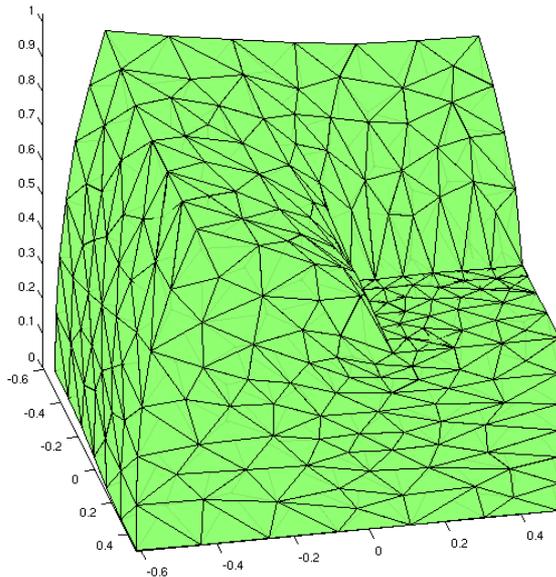
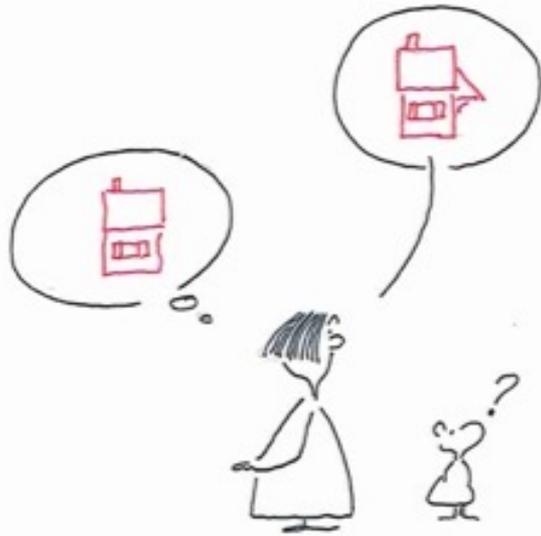


Elasticité linéaire

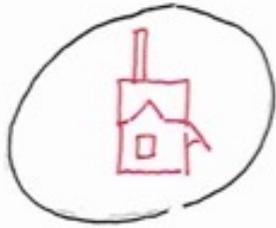
Y AURA DES
NOTES SUR LE COURS
... LE SOIR
C'EST PROMIS



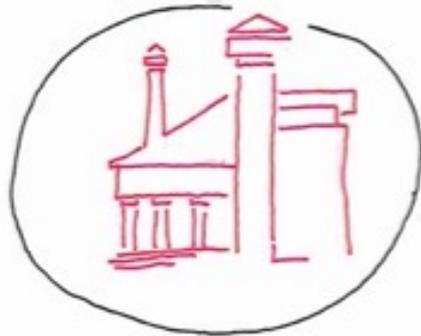
Le projet



J-15



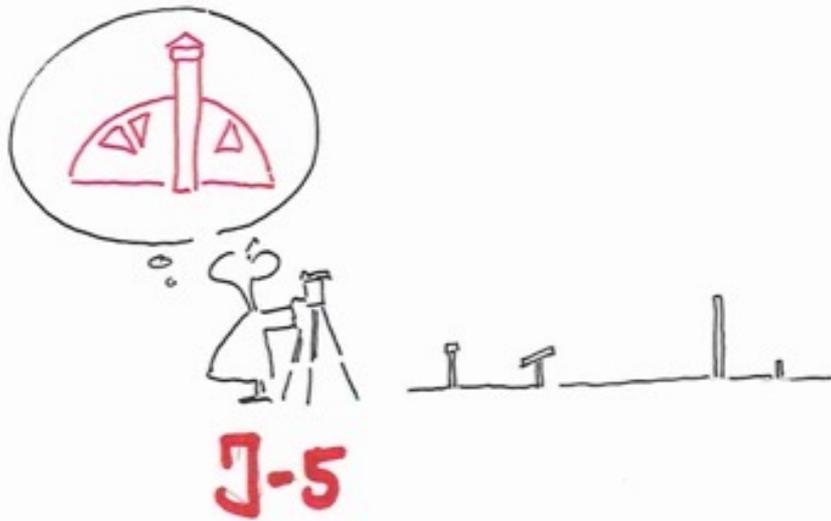
J-14



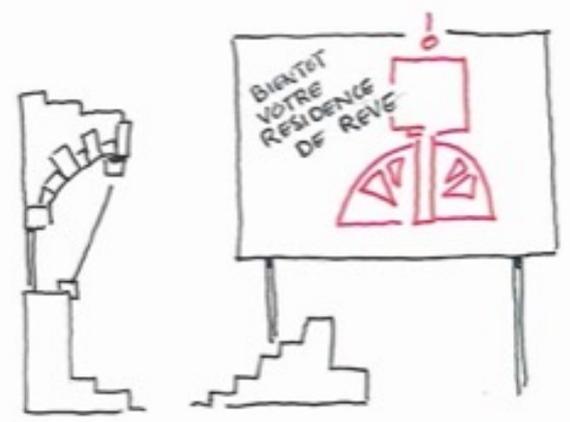
J-12

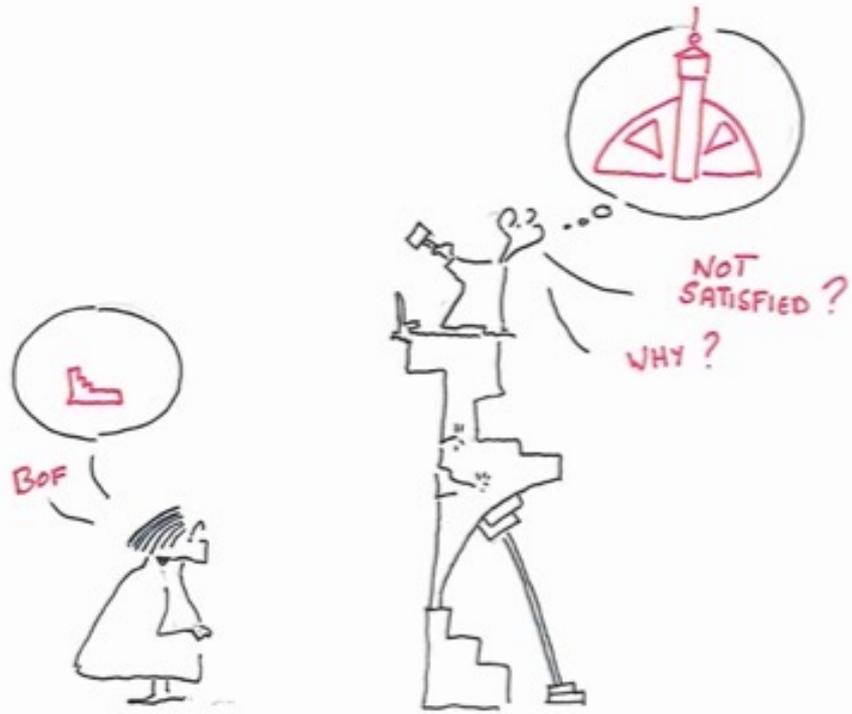


J-10



J-4





J

PROBLÈME ELLIPTIQUE



ELEMENTS FINIS *

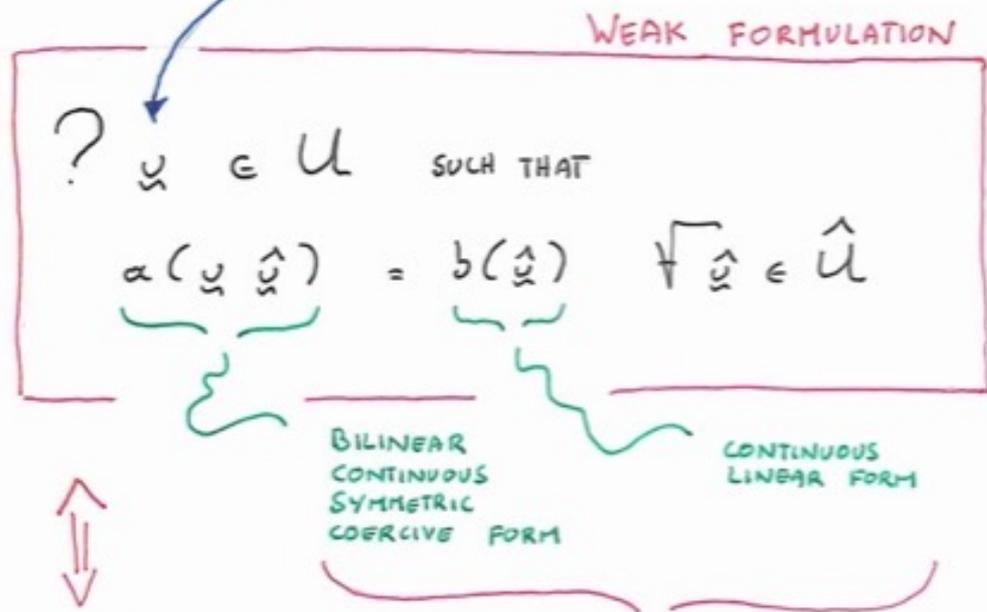


* "USUAL FEM"
CLASSICAL
GALERKIN
FORMULATION

ABSTRACT GENERIC ELLIPTIC PROBLEM

HEAT CONDUCTION
 LINEAR ELASTICITY
 SIMPLIFIED MODELS OF LINEAR ELASTICITY
 BEAM / SHELLS
 ROPE / MEMBRANE
 STOKES PROBLEM

Now,
 WE CONSIDER
 A VECTORIAL UNKNOWN FIELD !



ASSUMPTIONS
 REQUIRED
 TO HAVE AN ABSTRACT
 MINIMIZATION
 PROBLEM

? $\underline{u} \in \mathcal{U}$ SUCH THAT

$$\mathcal{J}(\underline{u}) = \min_{\underline{v} \in \mathcal{U}} \underbrace{\frac{1}{2} a(\underline{v}, \underline{v}) - b(\underline{v})}_{\mathcal{J}(\underline{v})}$$

MINIMIZATION
 PROBLEM

HEAT CONDUCTION

CONSERVATION LAW

$$-\nabla \cdot \underline{q} + f = 0$$

HEAT FLOW

CONSTITUTIVE LAW

$$\underline{q} = -k \nabla u$$

FOURIER

TEMPERATURE

? u SUCH THAT

$$\nabla \cdot (\overbrace{-\underline{q}(u)}^{-k \nabla u}) + f = 0 \quad \text{in } \Omega$$

$$-\underline{q} \cdot \underline{n} = g \quad \text{ON } \Gamma_2$$

$$u = 0 \quad \text{ON } \Gamma_0$$

3D LINEAR ISOTROPIC ELASTICITY



HOOKE ELASTIC BODY
SMALL DEF.

CONSTITUTIVE LAW

$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{1}}$$

CONSERVATION LAWS

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}} = 0$$

$$\underline{\underline{\sigma}} = \underline{\underline{\sigma}}^T$$

DEFORMATION TENSOR

$$\underline{\underline{\epsilon}} \triangleq \frac{1}{2} (\nabla \underline{\underline{u}} + (\nabla \underline{\underline{u}})^T)$$

?

$$\nabla \cdot \underline{\underline{\sigma}}(\underline{\underline{u}}) + \underline{\underline{f}} = 0 \quad \text{in } \Omega$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = \underline{\underline{g}} \quad \text{ON } \Gamma_2$$

$$\underline{\underline{u}} = 0 \quad \text{ON } \Gamma_0$$

STOKES PROBLEM

CONSERVATION LAWS

$$\cancel{(\underline{v} \cdot \nabla) \underline{v}} = \nabla \cdot \underline{\underline{b}} + \underline{f}$$

$$\nabla \cdot \underline{v} = 0$$

CREEPING FLOW
 $Re \ll 0$

INCOMPRESSIBLE FLOW

CONSTITUTIVE LAW

NEWTONIAN FLUID

$$\underline{\underline{\sigma}} = 2\mu \underline{\underline{d}} - p \underline{\underline{I}}$$

RATE OF DEFORMATION TENSOR

$$\underline{\underline{d}} \triangleq \frac{1}{2} (\nabla \underline{v} + (\nabla \underline{v})^T)$$

? (\underline{v}, p)

$$\left\{ \begin{array}{l} \nabla \cdot (\underline{\underline{b}}(\underline{v}, p)) + \underline{f} = 0 \\ \nabla \cdot \underline{v} = 0 \end{array} \right. \quad \text{in } \Omega$$

$$\underline{\underline{b}} \cdot \underline{e}_3 = \underline{\underline{\sigma}} \quad \text{on } \sqrt{2}$$

$$\underline{v} = 0 \quad \text{on } \sqrt{0}$$

CALCULUS

$$\nabla \cdot \underline{v} = 0$$



$\exists \psi$ SUCH THAT

$$\underline{v} \in \Omega$$

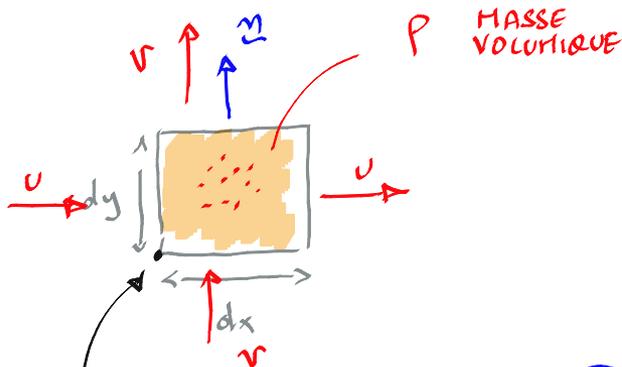
$$\underline{v} = \nabla \times \psi$$

STREAM
FUNCTION

? ψ

$$\nabla^4 \psi + \nabla \times \underline{f} = 0 \quad \text{in } \Omega$$

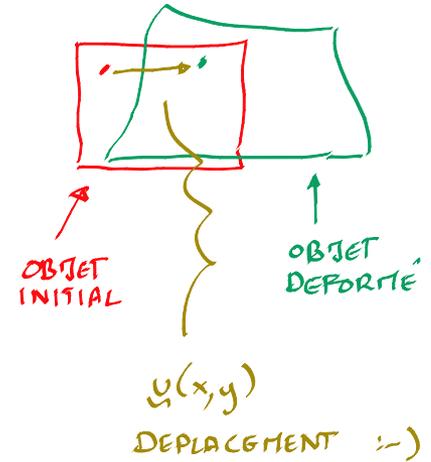
+ SUITABLE
BOUNDARY
CONDITIONS



$$\int_{\Omega} \nabla \cdot (\rho \underline{v})$$

$$\frac{d}{dt} \int_{\Omega} \rho \, dx \, dy = \int_{\partial \Omega} \underbrace{\rho \underline{v} \cdot \underline{n}}_{\text{VITESSE}}$$

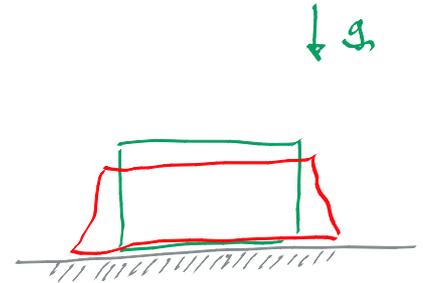
$$\frac{\partial \rho}{\partial t} = -\nabla \cdot (\rho \underline{v})$$

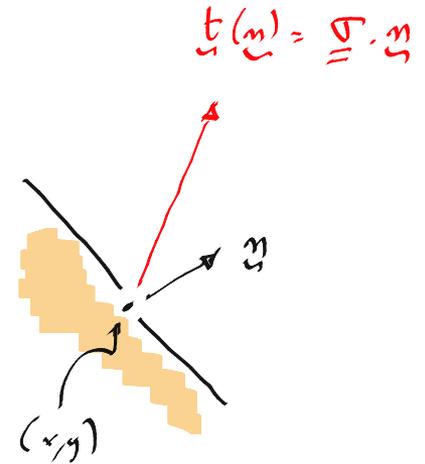
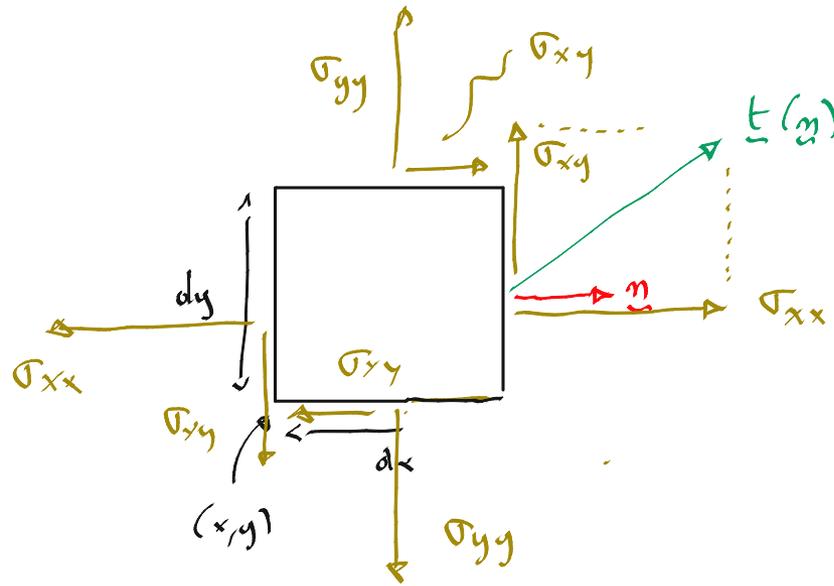


$$dx \, dy \, \frac{\partial \rho}{\partial t} = \underbrace{\left[\rho u(x, y) - \rho u(x + dx, y) \right]}_{\text{horizontal flux}} dy + \underbrace{\left[\rho v(x, y) - \rho v(x, y + dy) \right]}_{\text{vertical flux}} dx$$

Bilan
de masse...

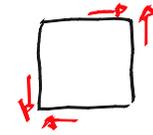
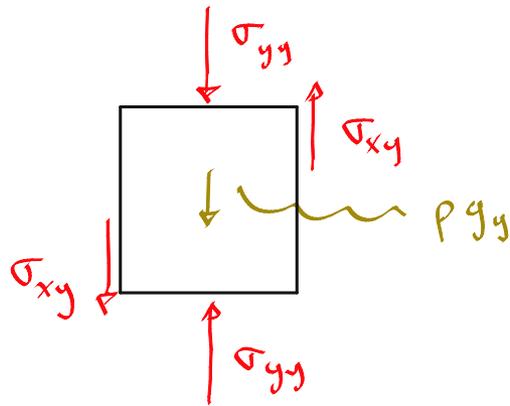
$$\frac{\partial \rho}{\partial t} = \frac{\partial}{\partial x} (\rho u) + \frac{\partial}{\partial y} (\rho v)$$





$$\underline{t}(\underline{n}) = \begin{bmatrix} \sigma_{xx} & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} \end{bmatrix} \cdot \begin{bmatrix} n_x \\ n_y \end{bmatrix}$$

Tenseur
de contraintes



$$\rho g_y + \frac{\partial}{\partial x}(\sigma_{xy}) + \frac{\partial}{\partial y}(\sigma_{yy}) = 0$$

$$\frac{\partial}{\partial x}(\sigma_{xx}) + \frac{\partial}{\partial y}(\sigma_{yx}) = 0$$

$$\nabla \cdot \underline{\underline{\sigma}} + \rho \underline{\underline{g}} = 0$$

Bilan
de quantité de mouvement

3D LINEAR ISOTROPIC ELASTICITY



HOOKE ELASTIC BODY
SMALL DEF.

CONSTITUTIVE LAW

$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{1}}$$

CONSERVATION LAWS

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}} = 0$$

$$\underline{\underline{\sigma}} = \underline{\underline{\sigma}}^T$$

DEFORMATION TENSOR

$$\underline{\underline{\epsilon}} \triangleq \frac{1}{2} (\nabla \underline{\underline{u}} + (\nabla \underline{\underline{u}})^T)$$

?

$$\nabla \cdot \underline{\underline{\sigma}}(\underline{\underline{u}}) + \underline{\underline{f}} = 0 \quad \text{in } \Omega$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = \underline{\underline{g}} \quad \text{ON } \Gamma_2$$

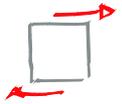
$$\underline{\underline{u}} = 0 \quad \text{ON } \Gamma_0$$

$$t_1(\underline{\underline{\epsilon}}) = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = \nabla \cdot \underline{\underline{u}}$$

$$\begin{bmatrix} \sigma_{xx} & & \\ \sigma_{xy} & \sigma_{yy} & \\ & & \end{bmatrix}$$

CONTRAINTES NORTALES

CONTRAINTE DE CISAILLEMENT



$$\begin{bmatrix} \epsilon_{xx} & & \\ \epsilon_{xy} & \epsilon_{yy} & \\ & & \end{bmatrix} = \frac{1}{2} \begin{bmatrix} 2\partial u/\partial x & & \\ \partial v/\partial x + \partial u/\partial y & & \\ & & 2\partial v/\partial y \end{bmatrix}$$

$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu \text{ SHEAR MODULUS}} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} t_1(\underline{\underline{\epsilon}}) \underline{\underline{S}}$$

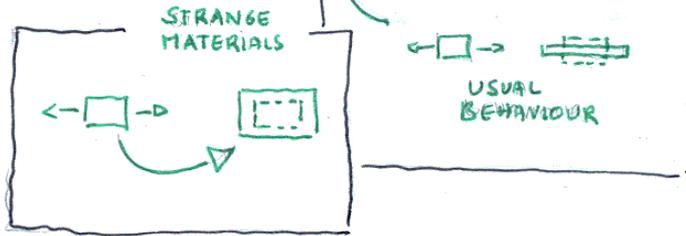
POISSON'S COEFFICIENT

$$\nu \in]-1;$$

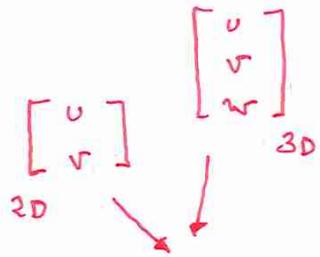
$$\frac{1}{2} [$$

INCOMPRESSIBLE MATERIAL

$$\lambda \rightarrow \infty \text{ IF } \nu \rightarrow \frac{1}{2}$$



ABSTRACT GENERIC DISCRETE FORMULATION



$$\underbrace{u(x)}_{\in U}$$

$$u^h(x) = \sum_{i=1}^m u_i \tau_i(x)$$

$$\in U^h \subset U$$

$$\dim(U^h) = 2m$$

$$\dots = 3m$$

3D PROBLEMS

? u^h SUCH THAT

$$a(u^h, \hat{u}^h) = b(\hat{u}^h)$$

$$\forall \hat{u}^h \in \hat{U}^h$$

DISCRETE
FORMULATION

$$U^h = \text{SPAN} \left\{ \underbrace{\begin{bmatrix} \tau_1 \\ 0 \end{bmatrix} \begin{bmatrix} \tau_2 \\ 0 \end{bmatrix} \dots \begin{bmatrix} \tau_m \\ 0 \end{bmatrix}}_m \underbrace{\begin{bmatrix} 0 \\ \tau_1 \end{bmatrix} \dots \begin{bmatrix} 0 \\ \tau_m \end{bmatrix}}_m \right\}$$

IS $\underline{\underline{\sigma}}(\underline{\hat{u}}, \underline{u})$
 SYMMETRIC?

$$\sigma_{ij} = C_{ijkl} \epsilon_{kl}$$

$$\langle \underline{\underline{\nabla}} \underline{\hat{u}} : \underline{\underline{\nabla}} \underline{u} \rangle = \langle \underbrace{\left(\frac{\underline{\underline{\nabla}} \underline{\hat{u}} + (\underline{\underline{\nabla}} \underline{\hat{u}})^T}{2} \right)}_{\text{SYM}} + \underbrace{\left(\frac{\underline{\underline{\nabla}} \underline{\hat{u}} - (\underline{\underline{\nabla}} \underline{\hat{u}})^T}{2} \right)}_{\text{ANTI-SYM}} : \underbrace{\underline{\underline{\nabla}} \underline{u}}_{\text{SYM}} \rangle$$



$$= \langle \underline{\underline{\epsilon}}(\underline{\hat{u}}) : \underline{\underline{\nabla}} \underline{u} \rangle$$



$$\underline{\underline{C}} : \underline{\underline{\epsilon}}(\underline{u})$$

GENERALIZED HOOKE'S LAW

$$= \langle \underline{\underline{\epsilon}}(\underline{\hat{u}}) : \underline{\underline{C}} : \underline{\underline{\epsilon}}(\underline{u}) \rangle$$

WEAK FORMULATION

? $u \in U$ SUCH THAT

$$\underbrace{\langle \underline{\underline{u}}(\hat{u}) : \underline{\underline{v}} : \underline{\underline{u}}(u) \rangle}_{a(\hat{u}, u)} = \underbrace{\langle \hat{u} \cdot f \rangle + \langle \hat{u} \cdot g \rangle_N}_{b(\hat{u})}$$

$\forall \hat{u} \in \hat{U}$

FOR SUITABLE ONLY $\underline{\underline{v}}!$

? $u \in U$ SUCH THAT

$$J(u) = \min_{v \in U} \underbrace{\frac{1}{2} a(u, v) - b(v)}_{J(v)}$$

MINIMIZATION PROBLEM

$$\frac{1}{2} \alpha(\underline{v}, \underline{v}) = \left\langle \frac{1}{2} \lambda \underline{\underline{\epsilon}}(\underline{v}) : \underline{\underline{\epsilon}}(\underline{v}) + \mu \underline{\underline{\epsilon}}(\underline{v}) : \underline{\underline{\epsilon}}(\underline{v}) \right\rangle$$

ENERGY
OF
DEFORMATION

MUST BE
A QUADRATIC POSITIVE FORM
IN ORDER TO OBTAIN A MINIMIZATION
PROBLEM

$$\begin{array}{l} \mu > 0 \\ \frac{3}{2} \lambda + \mu > 0 \end{array}$$

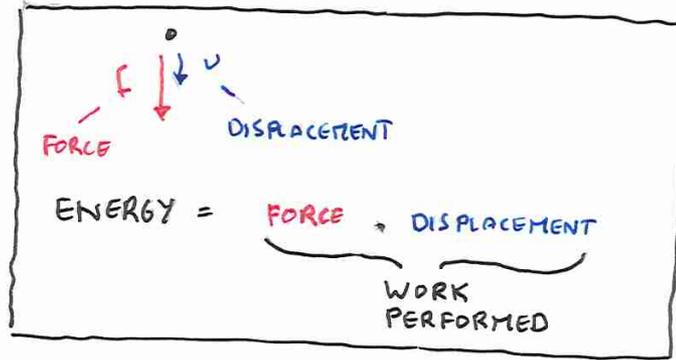
$$(\epsilon_{12} \neq 0)$$

$$(\epsilon_{11} = \epsilon_{22} = \epsilon_{33} \neq 0)$$

ADMISSIBLE VALUES
FOR LAMÉ COEFFICIENTS

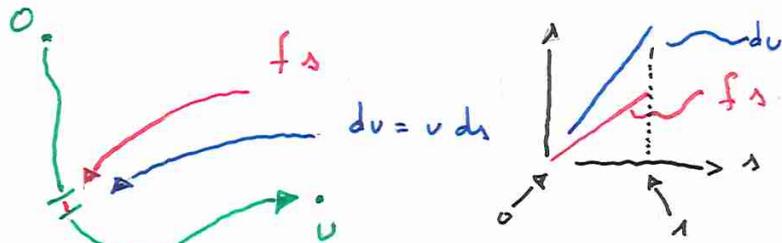
"DEFORMATION ENERGY"

WHAT IS IT?



ENERGY OF DEFORMATION

= WORK PERFORMED TO DEFORM THE STRUCTURE AND IS STORED INSIDE. SUCH AN ENERGY IS RECOVERED WHEN EXTERNAL FORCES ARE REMOVED



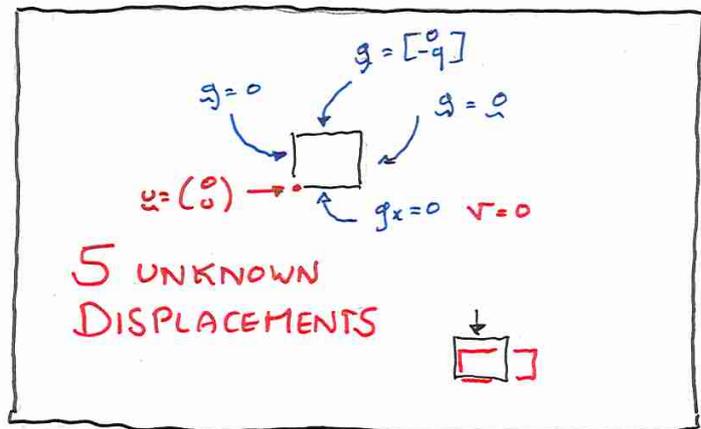
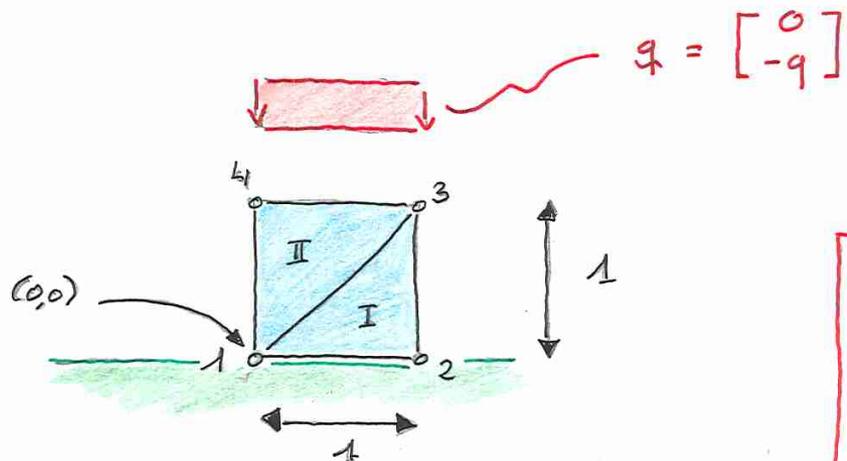
WORK PERFORMED

$$\begin{aligned}
 &= \int_0^u f \, du \\
 &= \int_0^1 f u \, ds \\
 &= f u \int_0^1 ds \\
 &= \frac{1}{2} f u !
 \end{aligned}$$

VERY VERY SLOW LOADING
 → QUASI-STATIC APPROACH
 • IRREVERSIBILITY } ARE NEGLECTED!
 • DISSIPATION

NUMERICAL EXAMPLE

$$\underline{\underline{c}} = \begin{bmatrix} 0 \\ \nu 0 \end{bmatrix}$$



$$? \underline{\underline{u}}^h = \sum_{i=1}^m \underline{\underline{U}}_i \tau_i$$

$$\langle \underline{\underline{\epsilon}}(\underline{\underline{u}}^h) : \underline{\underline{\sigma}}(\underline{\underline{u}}^h) \rangle = \langle \underline{\underline{f}} \cdot \underline{\underline{u}}^h \rangle + \langle \langle \underline{\underline{g}} \cdot \underline{\underline{u}}^h \rangle \rangle_N$$

$\begin{bmatrix} U_i \\ V_i \end{bmatrix}$

$$\begin{bmatrix} \tau_i \\ 0 \end{bmatrix} \quad \begin{bmatrix} 0 \\ \tau_i \end{bmatrix}$$

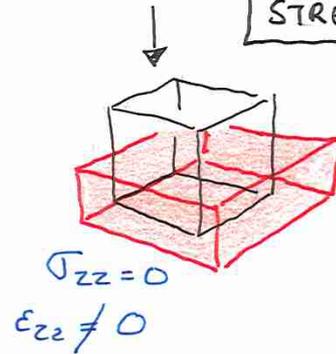
2m TEST FUNCTIONS

ANALYTICAL SOLUTION

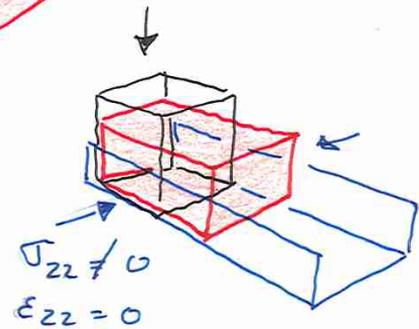
PLANAR STRESSES

$$\begin{aligned} \sigma_c &= \frac{q}{E} \begin{bmatrix} \nu x \\ -y \end{bmatrix} \\ \epsilon &= \frac{q}{E} \begin{bmatrix} \nu & 0 \\ 0 & -1 \end{bmatrix} \end{aligned}$$

PLANAR STRESSES



PLANAR DEFORMATIONS



$$\left. \begin{aligned} \Delta \cdot \epsilon_{zz} &= 0 \\ \epsilon_{zz} &= 0 \end{aligned} \right\} = 0!$$

$$\begin{aligned} \sigma_{xx} &= \frac{E}{(1-\nu^2)} \left(\underbrace{\epsilon_{xx}}_{\frac{q\nu}{E}} + \nu \underbrace{\epsilon_{yy}}_{-\frac{q}{E}} \right) = 0 \\ \sigma_{yy} &= \frac{E}{(1-\nu^2)} \left(\nu \underbrace{\epsilon_{xx}}_{\frac{q\nu}{E}} - \underbrace{\epsilon_{yy}}_{-\frac{q}{E}} \right) = -q! \end{aligned}$$

$$\begin{bmatrix} \tau_i \\ 0 \end{bmatrix} \text{ OR } \begin{bmatrix} 0 \\ \tau_i \end{bmatrix}$$

$$\langle \underbrace{\|m\| \begin{pmatrix} \tau_i \\ 0 \end{pmatrix}}_{\text{input}} : \underbrace{\|q\| \begin{pmatrix} \tau_i \\ 0 \end{pmatrix}}_{\text{output}} \rangle$$

$$\sum_j U_j \begin{bmatrix} \tau_j \\ 0 \end{bmatrix} + V_j \begin{bmatrix} 0 \\ \tau_j \end{bmatrix}$$

$$\sum_j \underline{A}_{ij} \cdot U_j$$

$$\sum_j \begin{bmatrix} U_j \\ V_j \end{bmatrix} \tau_j = \sum_j U_j \begin{bmatrix} \tau_j \\ 0 \end{bmatrix} + V_j \begin{bmatrix} 0 \\ \tau_j \end{bmatrix}$$

$$\sum_j \begin{bmatrix} \langle \|m\| \begin{pmatrix} \tau_i \\ 0 \end{pmatrix} : \|q\| \begin{pmatrix} \tau_j \\ 0 \end{pmatrix} \rangle & \langle \|m\| \begin{pmatrix} \tau_i \\ 0 \end{pmatrix} : \|q\| \begin{pmatrix} 0 \\ \tau_j \end{pmatrix} \rangle \\ \langle \|m\| \begin{pmatrix} 0 \\ \tau_i \end{pmatrix} : \|q\| \begin{pmatrix} \tau_j \\ 0 \end{pmatrix} \rangle & \langle \|m\| \begin{pmatrix} 0 \\ \tau_i \end{pmatrix} : \|q\| \begin{pmatrix} 0 \\ \tau_j \end{pmatrix} \rangle \end{bmatrix} \cdot \begin{bmatrix} U_j \\ V_j \end{bmatrix}$$

LH

DISCRETE OPERATOR

$$\underline{\underline{\sigma}} \begin{pmatrix} \tau_{i,j} \\ 0 \end{pmatrix} = \begin{bmatrix} \tau_{i,x} & \tau_{i,y}/2 \\ \tau_{i,y}/2 & 0 \end{bmatrix}$$

$$\underline{\underline{\sigma}} \begin{pmatrix} 0 \\ \tau_{i,j} \end{pmatrix} = \begin{bmatrix} 0 & \tau_{i,x}/2 \\ \tau_{i,x}/2 & \tau_{i,y} \end{bmatrix}$$

$$\underline{\underline{G}} \begin{pmatrix} \underline{\underline{\epsilon}} \end{pmatrix} = \begin{bmatrix} A \epsilon_{xx} + B \epsilon_{yy} & 2C \epsilon_{xy} \\ 2C \epsilon_{xy} & A \epsilon_{yy} + B \epsilon_{xx} \end{bmatrix}$$

HOW
TO
CALCULATE
IT ?

$\frac{E(1-\nu)}{(1+\nu)(1-2\nu)}$	$\frac{E}{(1-\nu^2)}$	A
$\frac{E\nu}{(1+\nu)(1-2\nu)}$	$\frac{E\nu}{(1-\nu^2)}$	B
	$\frac{E}{2(1+\nu)}$	C

PLANAR
DEFORMATIONS

PLANAR
STRESSES

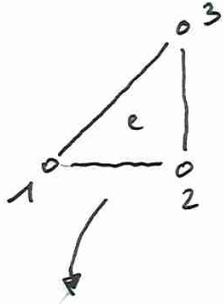
$$\underline{\underline{\epsilon}} : \underline{\underline{\sigma}} = \epsilon_{xx} (A \epsilon_{xx} + B \epsilon_{yy}) + 4C \epsilon_{xy} \epsilon_{xy} + \epsilon_{yy} (A \epsilon_{yy} + B \epsilon_{xx})$$

$$\underline{\underline{A}}_{ij} = \left[\begin{array}{l} \langle \tau_{i,x} A \tau_{j,x} \rangle + \langle \tau_{i,y} C \tau_{j,y} \rangle \\ \langle \tau_{i,y} B \tau_{j,x} \rangle + \langle \tau_{i,x} C \tau_{j,y} \rangle \\ \langle \tau_{i,x} B \tau_{j,y} \rangle + \langle \tau_{i,y} C \tau_{j,x} \rangle \end{array} \right]$$

$$\underline{\underline{\epsilon}} \left(\begin{array}{c} 0 \\ \tau_i \end{array} \right) : \underline{\underline{\sigma}} \left(\underline{\underline{\epsilon}} \left(\begin{array}{c} \tau_j \\ 0 \end{array} \right) \right)$$

LOCAL ELASTICITY MATRIX

COMPUTING A LOCAL ELASTICITY MATRIX



	$\tau_{i,x}$	$\tau_{i,y}$
1	-1	0
2	1	-1
3	0	1

$A \langle \tau_{i,x} \tau_{j,x} \rangle$ $+ C \langle \tau_{i,y} \tau_{j,y} \rangle$	$B \langle \tau_{i,x} \tau_{j,y} \rangle$ $+ C \langle \tau_{i,y} \tau_{j,x} \rangle$
$B \langle \tau_{i,y} \tau_{j,x} \rangle$ $+ C \langle \tau_{i,x} \tau_{j,y} \rangle$	$A \langle \tau_{i,y} \tau_{j,y} \rangle$ $+ C \langle \tau_{i,x} \tau_{j,x} \rangle$

A^e	$A_{xx11} = A$
	$A_{yy11} = C$
	$A_{xy11} = 0$
	$A_{yx11} = 0$

$$A^e_{=y} =$$

<table border="1"> <tr> <td>A</td> <td>0</td> </tr> <tr> <td>0</td> <td>C</td> </tr> </table>	A	0	0	C	-A	B	0	-B
A	0							
0	C							
	C	-C	-C	0				
	A+C	-B-C	-C	B				
	-B-C	A+C	C	-A				
			C	0				
			0	A				

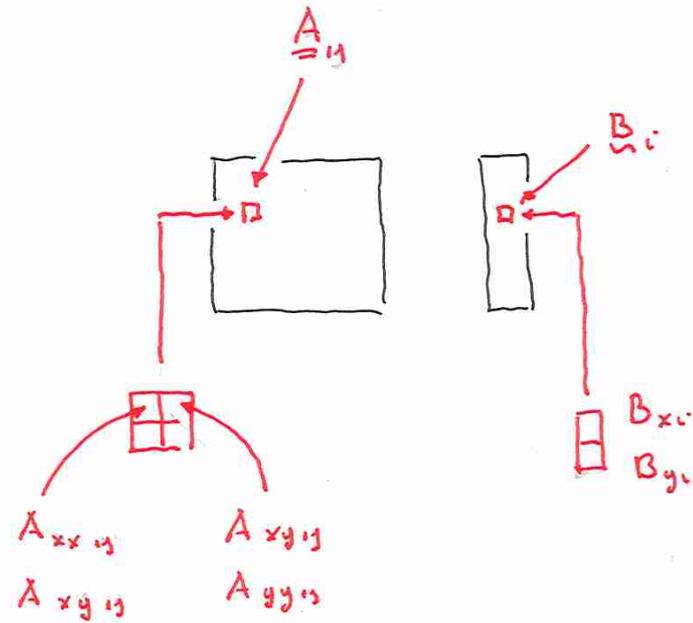
1/2

$$\underline{B}_i = \langle\langle g \cdot \hat{u}^h \rangle\rangle_N$$

$\left[\begin{matrix} \tau_i \\ 0 \end{matrix} \right]$ OR $\left[\begin{matrix} 0 \\ \tau_i \end{matrix} \right]$

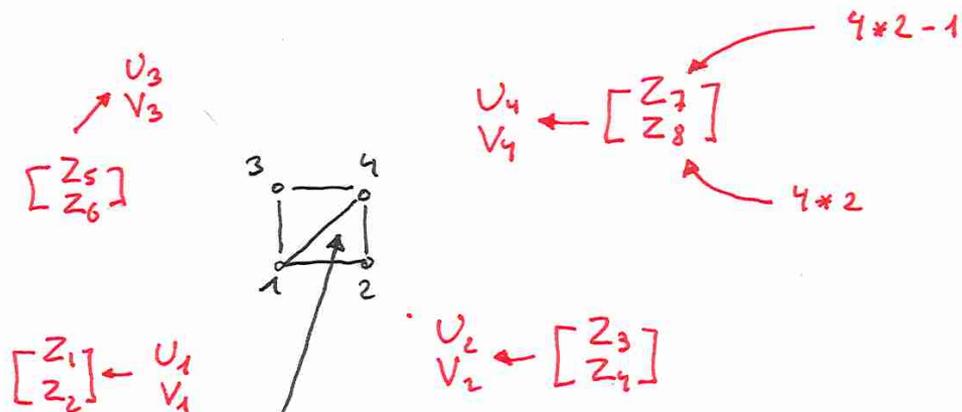
$$\begin{bmatrix} g_x \\ g_y \end{bmatrix}$$

$$\begin{bmatrix} \langle\langle g_x \tau_i \rangle\rangle \\ \langle\langle g_y \tau_i \rangle\rangle \end{bmatrix}$$



AND
HYPER
VECTOR !

ASSEMBLING PROCEDURE



$$\underline{A}_{ij}^e = [\quad]$$

6x6 MATRIX

$$\left[\begin{array}{c} \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \end{array} \right] \begin{bmatrix} Z_1 \\ \vdots \\ Z_8 \end{bmatrix} = \begin{bmatrix} \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \end{bmatrix}$$

GLOBAL HORIZONTAL INDEX

$$= \underbrace{2 * \text{GLOBAL SCALAR INDEX}}_{\text{GLOBAL}} - 1$$

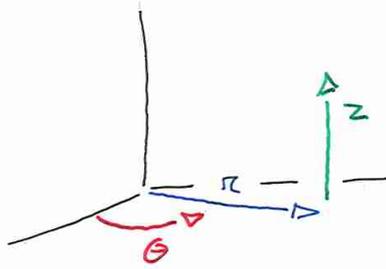
- $$\begin{bmatrix} U_1 \\ V_1 \\ U_2 \\ V_2 \\ U_3 \\ V_3 \\ U_4 \\ V_4 \end{bmatrix}$$

AXISYMMETRIC PROBLEMS

AXISYMMETRY!

$$v_\theta = 0$$

$$\frac{\partial}{\partial \theta} = 0$$



$$u = \begin{bmatrix} v_r \\ v_\theta \\ v_z \end{bmatrix} = \begin{bmatrix} v(r, z) \\ 0 \\ v(r, z) \end{bmatrix}$$

$$||m = \begin{bmatrix} v_{r,r} & 0 & (v_{r,z} + v_{z,r})/2 \\ v_r/r & 0 & 0 \\ 0 & 0 & v_{z,z} \end{bmatrix}$$

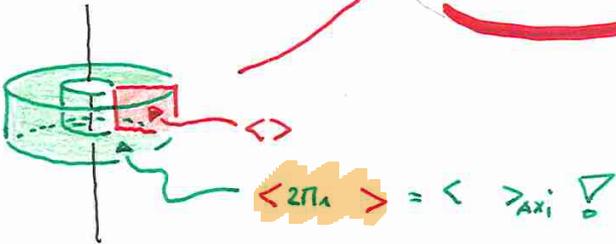
A, B, C
OF PLANAR
DEFORMATIONS !

$$||\epsilon = \begin{bmatrix} A \epsilon_{rr} + B (\epsilon_{\theta\theta} + \epsilon_{zz}) & 0 & 2C \epsilon_{rz} \\ A \epsilon_{\theta\theta} + B (\epsilon_{rr} + \epsilon_{zz}) & 0 & 0 \\ A \epsilon_{zz} + B (\epsilon_{rr} + \epsilon_{\theta\theta}) & 0 & 0 \end{bmatrix}$$

$$\mathbb{E}([\tau_i \ 0 \ 0]) = \begin{bmatrix} \tau_{i,2} & 0 & \tau_{i,2}/2 \\ & \tau_i/2 & 0 \\ & & 0 \end{bmatrix}$$

$$\mathbb{E}([0 \ 0 \ \tau_i]) = \begin{bmatrix} 0 & 0 & \tau_{i,2}/2 \\ & 0 & 0 \\ & & \tau_{i,2} \end{bmatrix}$$

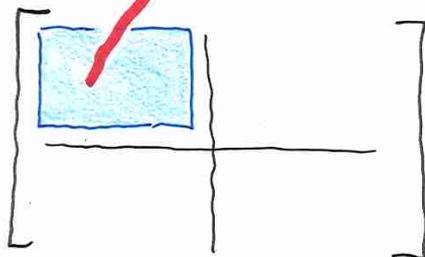
$$2\pi \left[\begin{array}{c} \langle \mathbb{E} \left(\begin{pmatrix} \tau_i \\ 0 \\ 0 \end{pmatrix} \right) : \mathbb{E} \left(\begin{pmatrix} \tau_{i,1} \\ 0 \\ 0 \end{pmatrix} \right) \rangle \\ \langle \mathbb{E} \left(\begin{pmatrix} 0 \\ 0 \\ \tau_i \end{pmatrix} \right) : \mathbb{E} \left(\begin{pmatrix} 0 \\ \tau_{i,1} \\ 0 \end{pmatrix} \right) \rangle \\ \hline \langle \mathbb{E} \left(\begin{pmatrix} \tau_i \\ 0 \\ 0 \end{pmatrix} \right) : \mathbb{E} \left(\begin{pmatrix} 0 \\ 0 \\ \tau_{i,2} \end{pmatrix} \right) \rangle \\ \langle \mathbb{E} \left(\begin{pmatrix} 0 \\ 0 \\ \tau_i \end{pmatrix} \right) : \mathbb{E} \left(\begin{pmatrix} 0 \\ \tau_{i,2} \\ 0 \end{pmatrix} \right) \rangle \end{array} \right]$$



\mathbb{A}_{ij}

$$\|m\| \begin{pmatrix} \tau_i \\ 0 \\ 0 \end{pmatrix} = \begin{bmatrix} \tau_{i,x} & 0 & \tau_{i,z}/2 \\ & \tau_i/r & 0 \\ & & 0 \end{bmatrix}$$

$$\|q\| \begin{pmatrix} \tau_j \\ 0 \\ 0 \end{pmatrix} = \begin{bmatrix} A\tau_{j,x} + B\tau_{j,z} & 0 & C\tau_{j,z} \\ & A\tau_{j,z} + B\tau_{j,x} & 0 \\ & & B\tau_{j,z} + B\tau_{j,x} \end{bmatrix}$$



$\|m\| \begin{pmatrix} \tau_i \\ 0 \\ 0 \end{pmatrix} : \|q\| \begin{pmatrix} \tau_j \\ 0 \\ 0 \end{pmatrix} = \langle \tau_{j,x} (A\tau_{j,x} + B\tau_{j,z}) r \rangle$
 $+ \langle \tau_{j,z} C\tau_{j,z} r \rangle$
 $+ \langle \frac{\tau_i}{r} (A\frac{\tau_{j,z}}{r} + B\tau_{j,x}) r \rangle$

$\underline{\underline{A}}_{ij}$

IT IS ALMOST
LIKE A 2D PROBLEM !

$$A \langle \tau_{i,\lambda} \tau_{j,\lambda} \rangle$$

$$+ C \langle \tau_{i,z} \tau_{j,z} \rangle$$

$$+ B \langle \tau_{i,\lambda} \tau_j \rangle$$

$$+ \langle \tau_i (B \tau_{j,\lambda} + A \frac{\tau_j}{r}) \rangle$$

ϵ_{00}/r

σ_{00}

$$B \langle \tau_{i,z} \tau_{j,\lambda} \rangle$$

$$+ C \langle \tau_{i,\lambda} \tau_{j,z} \rangle$$

$$+ B \langle \tau_{i,z} \tau_j \rangle$$

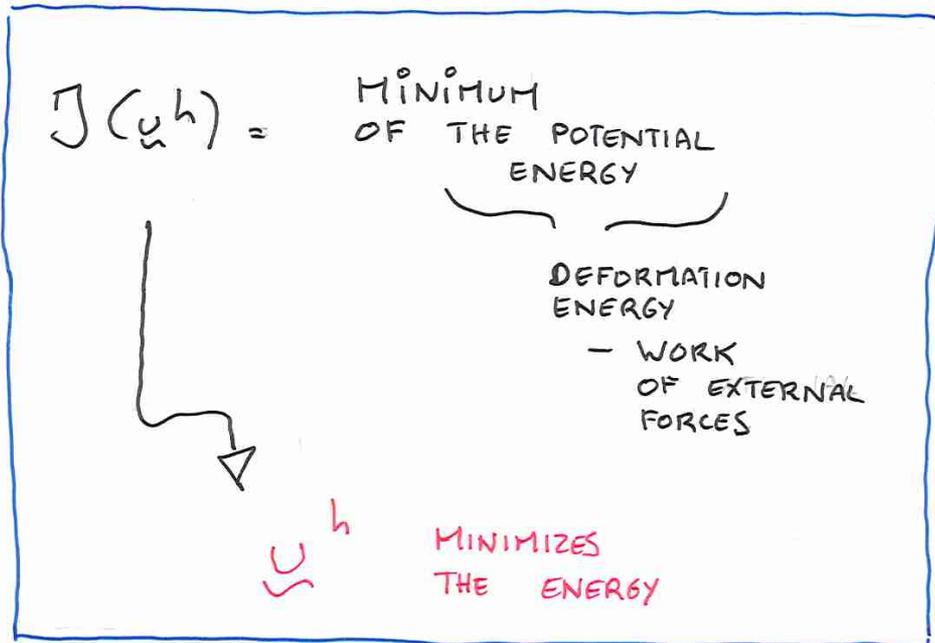
$$B \langle \tau_{i,\lambda} \tau_{j,z} \rangle$$

$$+ C \langle \tau_{i,z} \tau_{j,\lambda} \rangle$$

$$+ B \langle \tau_i \tau_{j,z} \rangle$$

$$A \langle \tau_{i,z} \tau_{j,z} \rangle$$

$$+ C \langle \tau_{i,\lambda} \tau_{j,\lambda} \rangle$$



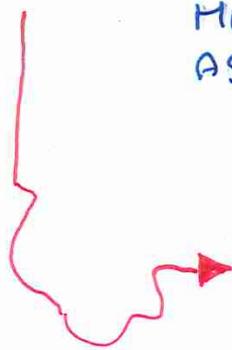
DISPLACEMENTS FORMULATION

DISCONTINUOUS! → SEVERAL VALUES AT EACH VERTEX!

TYPICALLY CONTINUOUS LINEAR OR QUADRATIC

BUT $\underline{\sigma}(u^h)$

MUST BE VIEWED AS A LEAST-SQUARE FIT OF THE EXACT SOLUTION $\underline{\sigma}(u)$



ALLOWS US TO USE THE BEST STRATEGY FOR THE INTERPRETATION OF $\underline{\sigma}(u^h)$

$G(u^h)$ = SOLUTION OF A LEAST-SQUARES FIT PROBLEM

$$a(\hat{u}, u) = b(\hat{u}) \\ \forall \hat{u} \in \hat{U}$$

$$a(\hat{u}^h, u^h) = b(\hat{u}^h) \\ \forall u^h \in \hat{U}^h$$

? u^h

$$a(\hat{u}^h, u^h - u) = 0 \\ \forall \hat{u}^h \in \hat{U}^h$$

CFR $\hat{u}^h \in \hat{U}$!
CFR LINEARITY!

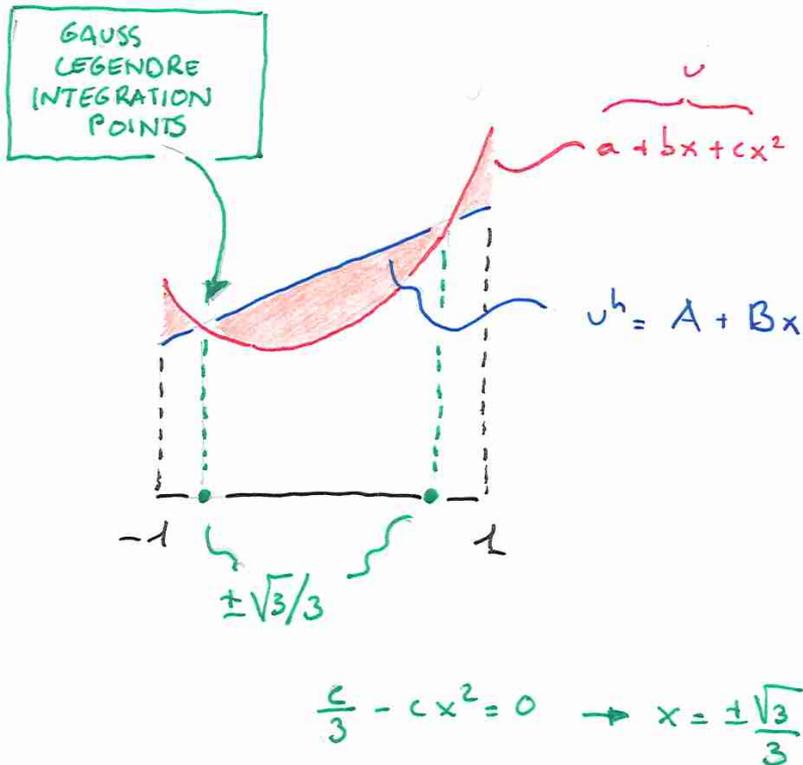
$\hat{b}(\hat{u}^h) = a(\hat{u}^h, u)$
 $\hat{a}(\hat{u}^h, u^h) = a(\hat{u}^h, u^h)$

? u^h

$$G(u^h) = \min_{v^h \in U^h} \frac{1}{2} \langle (\underline{\underline{\varepsilon}}(v^h) - \underline{\underline{\varepsilon}}(u)) : \underline{\underline{\varepsilon}} : (\underline{\underline{\varepsilon}}(v^h) - \underline{\underline{\varepsilon}}(u)) \rangle$$

$G(u^h)$

WHAT ARE THE "BEST VALUES" OF A LEAST-SQUARES POLYNOMIAL FIT ?



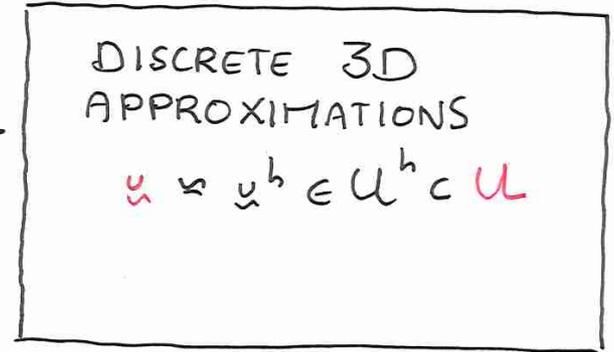
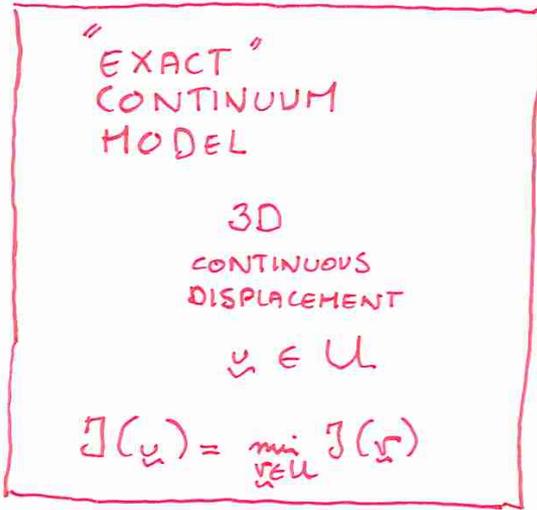
? A, B SUCH THAT

$$G(A, B) = \min_{\tilde{A}, \tilde{B}} \int_{-1}^1 (\tilde{A} + \tilde{B}x - a - bx - cx^2)^2 dx$$

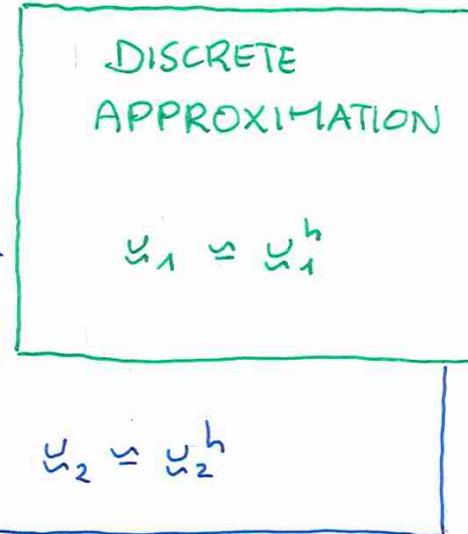
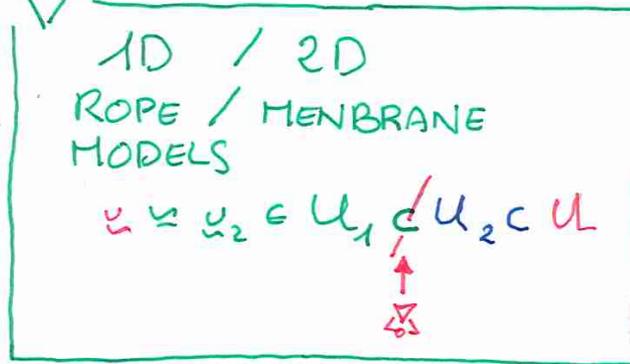
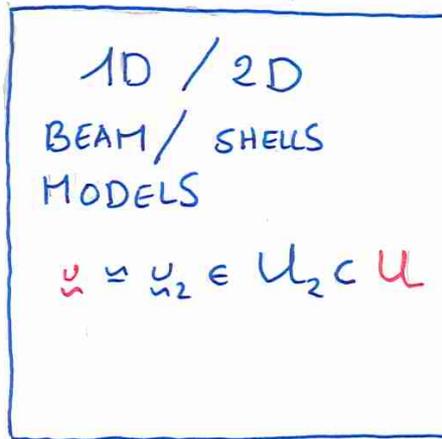
$$\begin{cases} B \int_{-1}^1 (A + Bx - a - bx - cx^2) x dx = 0 \\ A \int_{-1}^1 (A + Bx - a - bx - cx^2) dx = 0 \end{cases}$$

$$\begin{cases} B^2 \frac{2}{3} - Bb \frac{2}{3} = 0 \rightarrow B = b \\ 2A^2 - 2Aa - Ac \frac{2}{3} = 0 \rightarrow A = a + \frac{c}{3} \end{cases}$$

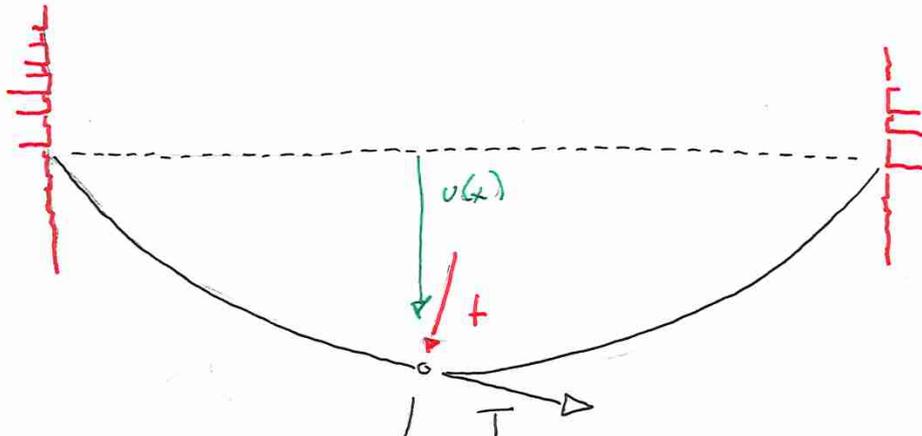
SIMPLIFIED MODELS



A HIERARCHICAL VIEW



ROPE MODEL

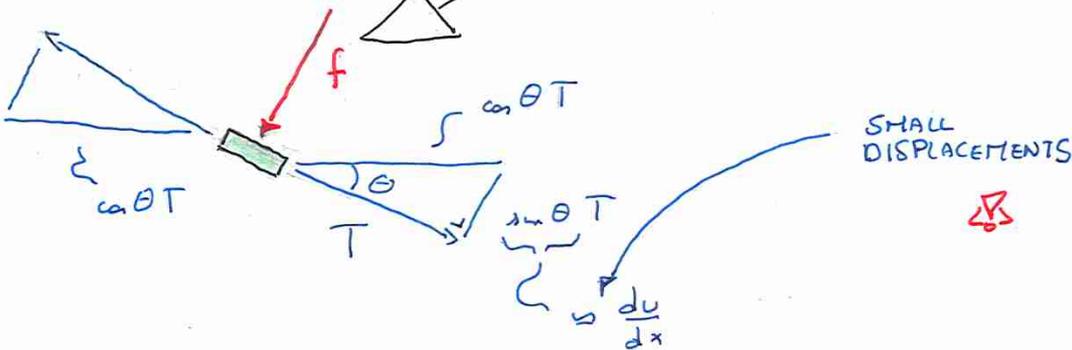


VERTICAL
FORCE
BALANCE

$$\frac{d}{dx} \left(T \frac{du}{dx} \right) = -f$$



$$T \frac{d^2 u}{dx^2} + f = 0$$



THEORY OF BEAMS

BENDING BEAM
• SMALL DEF
• ST VENANT PRINCIPLE
• NAVIER-BERNOULLI ASS.

FORCE & MOMENTUM
BALANCES

$$\frac{dQ}{dx} = -q$$

$$\frac{dM}{dx} = Q$$

CONSTITUTIVE
LAW

$$M = EI \frac{1}{R}$$

CURVATURE
OF BEAM

$$\frac{1}{R} = -\frac{d^2u}{dx^2}$$

?

$$\frac{d^4u}{dx^4} - \frac{q}{EI} = 0$$

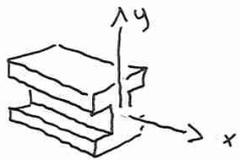
+ 4 BOUNDARY
CONDITIONS

$$\begin{aligned}
 & \left\langle EI \frac{d^4 u}{dx^4} \hat{u} \right\rangle = \langle q \hat{u} \rangle \\
 & - \left\langle EI \frac{d^3 u}{dx^3} \frac{d\hat{u}}{dx} \right\rangle = \langle q \hat{u} \rangle - \left[\overbrace{EI \frac{d^3 u}{dx^3}}^Q \hat{u} \right] \\
 & \left\langle EI \frac{d^2 u}{dx^2} \frac{d^2 \hat{u}}{dx^2} \right\rangle = \langle q \hat{u} \rangle - [Q \hat{u}] \\
 & \quad + \left[\underbrace{EI \frac{d^2 u}{dx^2}}_M \frac{d\hat{u}}{dx} \right] \\
 & \underbrace{\hspace{10em}}_{\alpha(u, \hat{u})} \qquad \underbrace{\hspace{10em}}_{\beta(\hat{u})}
 \end{aligned}$$

IMPOSING u OR Q
 $\frac{du}{dx}$ OR M
 AT BOTH ENDS IS EASY!

IT IS A MINIMIZATION PROBLEM ...

$$J(u) = \frac{1}{2} \left\langle EI \frac{d^2 u}{dx^2}, \frac{d^2 u}{dx^2} \right\rangle$$



$M(u)$
MOMENTUM

$\kappa(u)$
CURVATURE

DEFORMATION
ENERGY

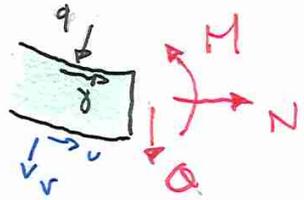
$$- \langle q, u \rangle + [Q, u] - [M, \frac{du}{dx}]$$

WORK
OF EXTERNAL
FORCES

$$\frac{1}{2} \int_0^L \int_S \sigma \epsilon \, dS \, dx$$

$E_y \frac{d^2 u}{dx^2}$ $y \frac{d^2 u}{dx^2}$

TIMOSHENKO BEAM THEORY



CONSERVATION
BALANCES

$$\frac{dN}{dx} = -\gamma$$

$$\frac{dQ}{dx} = -q$$

$$\frac{dM}{dx} = Q$$

CONSTITUTIVE
LAW

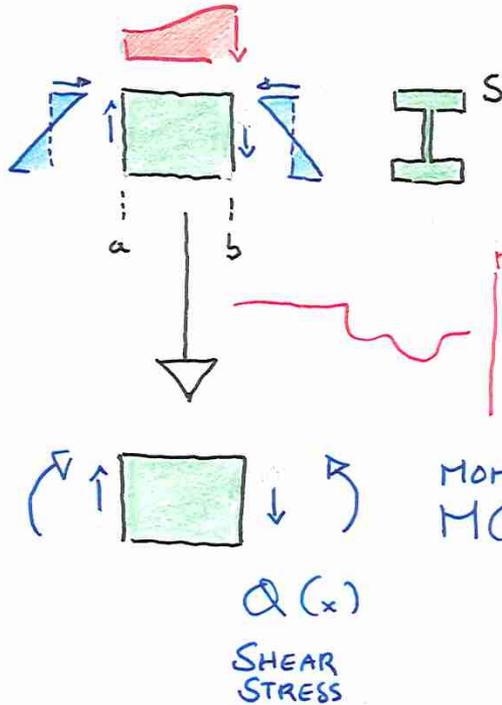
$$M = -EI \frac{d^2 v}{dx^2}$$

$$N = ES \frac{du}{dx}$$

$$EI \frac{d^4 v}{dx^4} = q$$

$$ES \frac{d^2 u}{dx^2} = -\gamma$$

BENDING OF A BEAM



THE PATTERN OF INTERNAL STRESS PRODUCES A TWISTING FORCE, A MOMENT ON THE MATERIAL

HORIZONTAL FORCE OBSERVED

$$\int_S (\sigma) dS = 0$$

MOMENTUM OBSERVED

$$\int_S (\sigma) \eta dS = M$$

HOOKE

$$= \int_S \frac{E \eta^2}{R} dS$$

$$= \frac{E}{R} \int_S \eta^2 dS$$

$$I = \frac{bh^3}{12}$$

$$= EI \frac{1}{R} = -EI \frac{d^2 u}{dx^2}$$

FORCE BALANCE

$$\int_a^b q \, dx - Q_a + Q_b = 0$$

$$\int_a^b q + \frac{dQ}{dx} \, dx = 0$$

$$\frac{dQ}{dx} = -q$$

$$\frac{dM}{dx} = Q$$

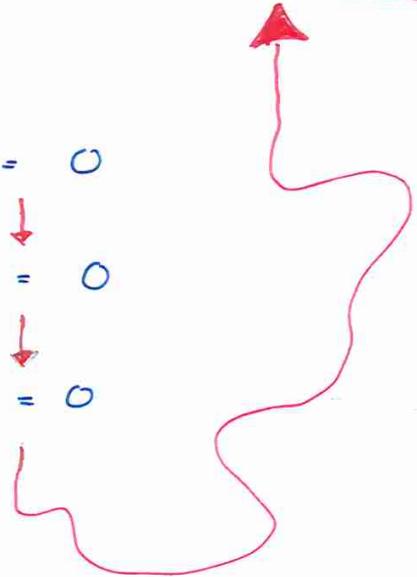
MOMENTUM BALANCE

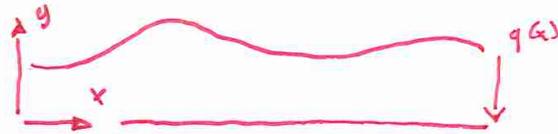
$$M_b - M_a - \int_a^b (x-a) q \, dx - (b-a) Q_b = 0$$

$$\int_a^b \left(\frac{dM}{dx} + (x-a) \frac{dQ}{dx} \right) dx - (b-a) Q_b = 0$$

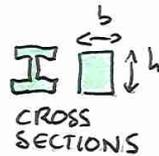
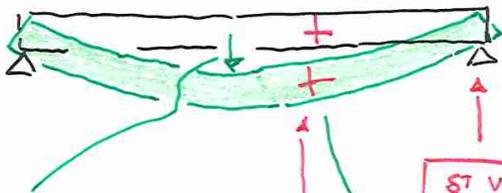
$$\int_a^b \left(\frac{dM}{dx} - Q \right) dx - (b-a) Q_b = 0$$

$$+ \left[(x-a) Q \right]_a^b$$





SMALL DEFORMATIONS



ST VENANT PRINCIPLE

NAVIER-BERNOULLI ASSUMPTION

UNKNOWN $v(x)$

LOCAL APPROXIMATION OF A CIRCULAR BEAM IS OK

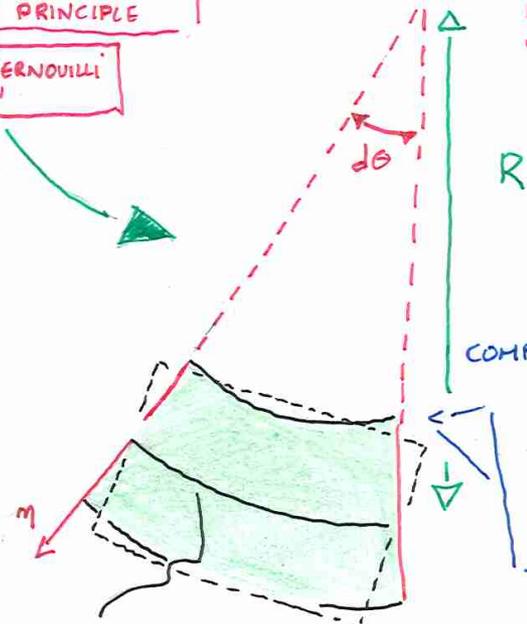
$$\epsilon = \frac{(R+m)d\theta - R d\theta}{R d\theta}$$

$$\downarrow$$

$$= \frac{m}{R}$$

$$\downarrow$$

$$= -\gamma \frac{dv}{dx^2}$$



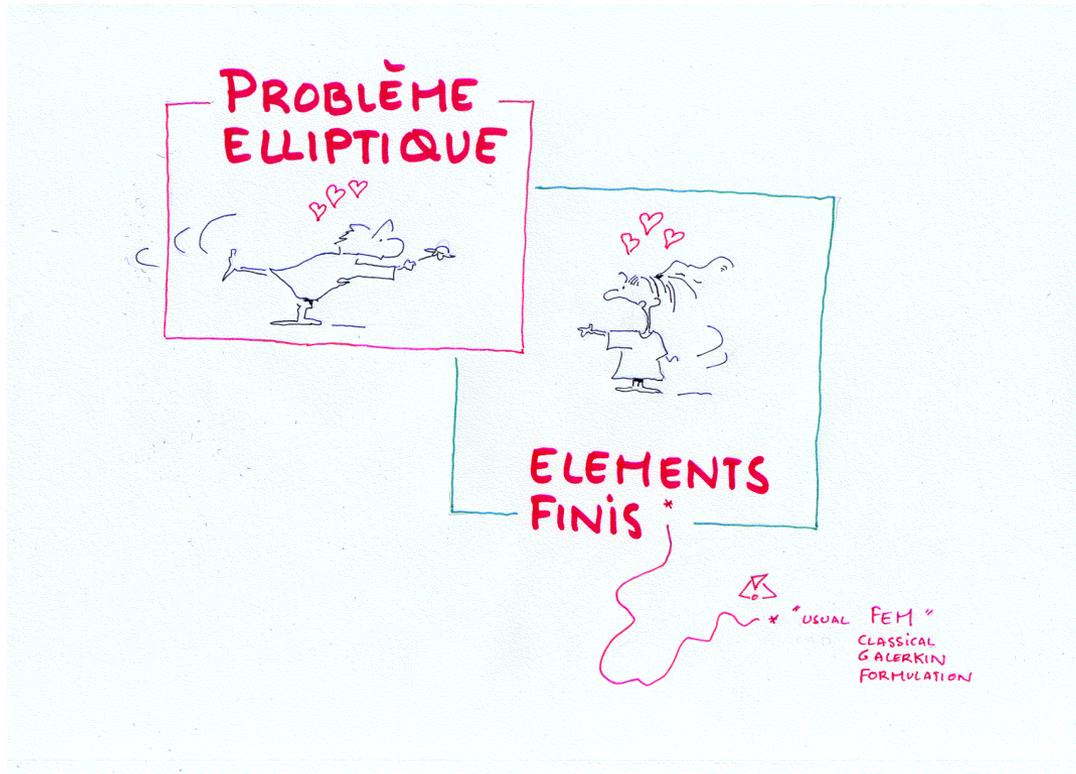
HOOKE

COMPRESSION

$$\sigma = E \epsilon$$

TRACTION

MIDPLANE ASSUMED UNSTRETCHED



Galerkin, c'est donc optimal
pour des équations elliptiques