

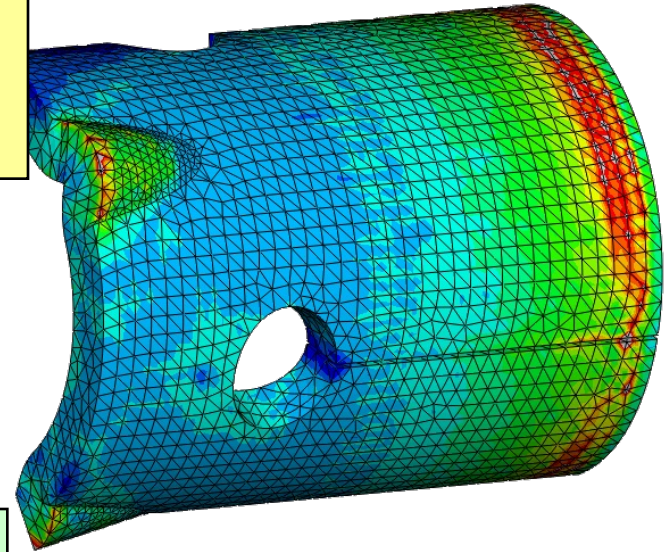
A quoi cela sert
les éléments finis ?

Antoine



Plan du cours et évaluation

Comment intégrer
numériquement
une fonction
sur un carré ?



Comment résoudre
l'élasticité linéaire ?

Evaluation continue

S1-S8 : 8 cours et 7 petits problèmes *A*

Evaluation certificative

S7-S9 : mini-projet *B*

S9-S12 : 4 cours théorie éléments finis

En session : examen écrit *C*

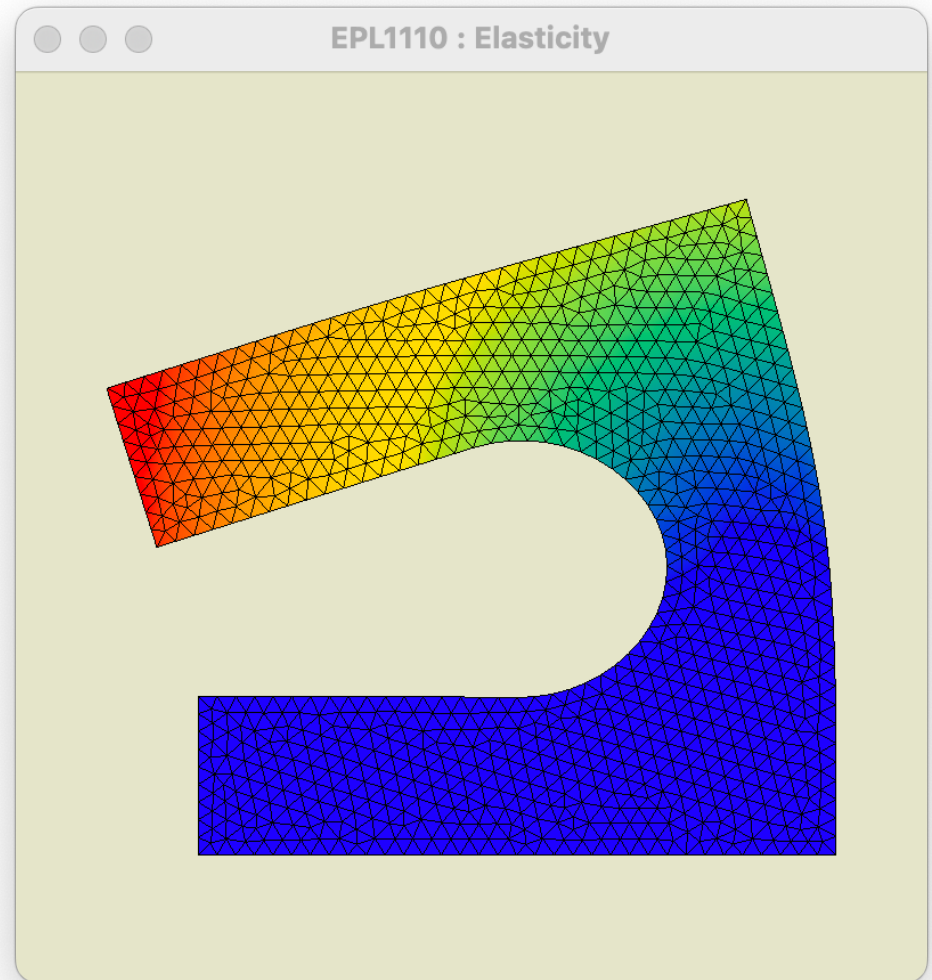
En juin, note finale $= (A+B+C)/3$ *si $(B+C)/2 > 10$*
 $= (B+C)/2$ *sinon*

En septembre, examen écrit

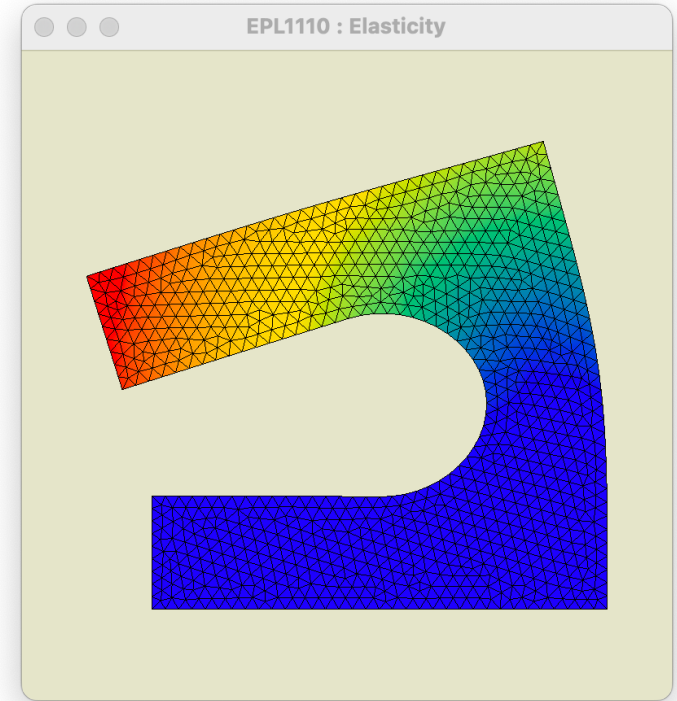
Pas de projet de rattrapage !

Evaluation

Elasticité linéaire



Hoping that
deadlines
are flexible ?



Définition du problème

Soumission du texte de deux pages
Approbation finale par l'assistants

Vendredi 15 mars
Vendredi 22 mars

Soumission du projet

Soumission du code du projet et du rapport

Vendredi ~~12~~ avril

19

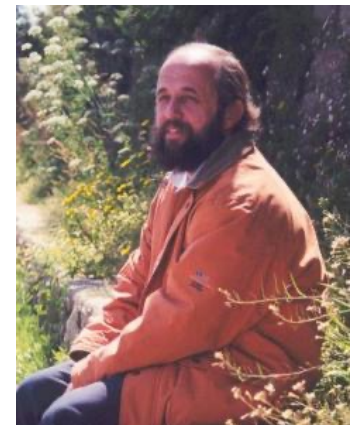
Interview sur le projet

Interview avec votre assistant de référence

Jeudi 18 avril
Vendredi 19 avril

23

26



10-08-10-10-00-00-00-00	No problem submitted :-(
10-10-10-08-00-00-00-00	File problem.pdf :-)
10-10-10-10-00-00-00-00	No problem submitted :-(
05-10-09-10-00-00-00-00	No problem submitted :-(
10-08-10-07-00-00-00-00	No problem submitted :-(
10-08-10-08-00-00-00-00	No problem submitted :-(
10-05-10-10-00-00-00-00	No problem submitted :-(
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10-10-04-05-00-00-00-00	No problem submitted :-(
05-10-10-10-00-00-00-00	File problem.pdf :-)
10-10-00-07-00-00-00-00	No problem submitted :-(
10-10-10-10-00-00-00-00	No problem submitted :-(
10-06-10-10-00-00-00-00	No problem submitted :-(
10-10-00-10-00-00-00-00	No problem submitted :-(
10-10-10-10-00-00-00-00	No problem submitted :-(
10-08-10-10-00-00-00-00	No problem submitted :-(
09-10-10-09-00-00-00-00	No problem submitted :-(
09-10-10-09-00-00-00-00	No problem submitted :-(

Votre
soumission
préliminaire...

1 Introduction

Pour le projet d'éléments finis, nous avons décidé d'analyser les forces appliquées par une personne qui est sur une échelle verticale.

2 Variantes

Notre problème de base concerne une personne qui est debout sur l'échelon d'une échelle classique telle que représentée à la figure 1 mais nous aimerions aussi proposer d'autres variantes:

- Notre personnage est placé sur des échellons à différents niveaux pour voir les différents cas de charge.
- Notre personnage est en train de monter sur l'échelle et ses mains appliquent également une force sur des échellons supérieurs à celui où ses pieds sont posés.
- Notre échelle a une forme différente, plus large ou avec des échellons de largeurs différentes.

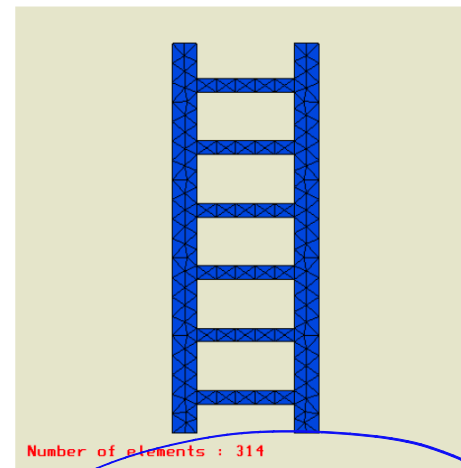


Figure 1: Résultats du filtre de Butterworth.

Votre soumission préliminaire...

2 Géométrie et Maillage

Notre maillage aurait la forme suivante :

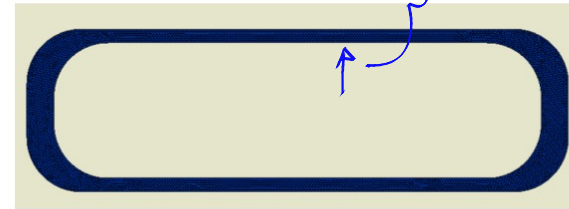


FIGURE 1 – Exemple du Maillage

Ce qui correspondrait à la géométrie de notre problème sur le schéma suivant :

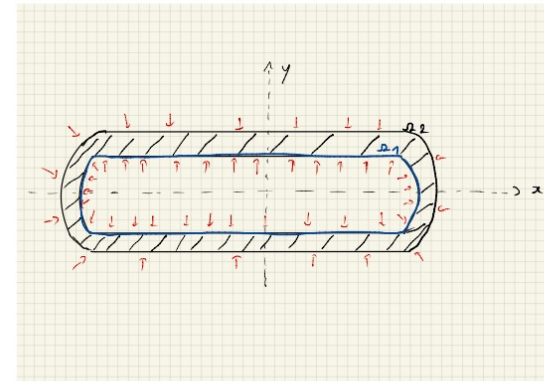


FIGURE 2 – Géométrie du problème

1. Sur Ω_1

2. Sur Ω_2

$$\sigma \cdot n = F_1 = p_1 * S_1$$

$\left[\frac{N}{m^2} \right]$ $[N]$ $[N/m^2]$
 $[m^2]$

$$\sigma \cdot n = F_2 = p_0 * S_2$$

où S_1 et S_2 représentent les surfaces à l'intérieur et à l'extérieur du réservoir, p_1 la pression intérieure (plus au moins 700 bars)^[1] et p_0 la pression atmosphérique évaluée à 1 bar. Il s'agit donc de conditions frontières de Neumann.

Nous poserons également les hypothèses suivantes :

1. répartition du gaz constante (forces appliquée constantes)
2. Problème axisymétrique

Nous devons naturellement détailler les surfaces et leur géométrie (approximation a 2 cylindres).

Il est à noter que puisque le problème contient des symétries nous pouvons réaliser une économie sur les calculs et le maillage en ne représentant qu'un quart du problème en imposant donc une condition de Dirichlet nul sur les surfaces de coupure sans perte de généralité. Nous pensions également rajouter une vanne de pression à la bonbonne afin de complexifier un peu le maillage mais il faudrait évaluer l'impact que cela aurait sur les équations et les conditions aux frontières.

3D LINEAR ISOTROPIC ELASTICITY



HOOKE ELASTIC BODY
SMALL DEF.

CONSTITUTIVE LAW

$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{S}}$$

CONSERVATION LAWS

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}} = 0$$

$$\underline{\underline{\sigma}} = \underline{\underline{\sigma}}^T$$

DEFORMATION TENSOR

$$\underline{\underline{\epsilon}} \triangleq \frac{1}{2} (\nabla_{\underline{\underline{u}}} + (\nabla_{\underline{\underline{u}}})^T)$$

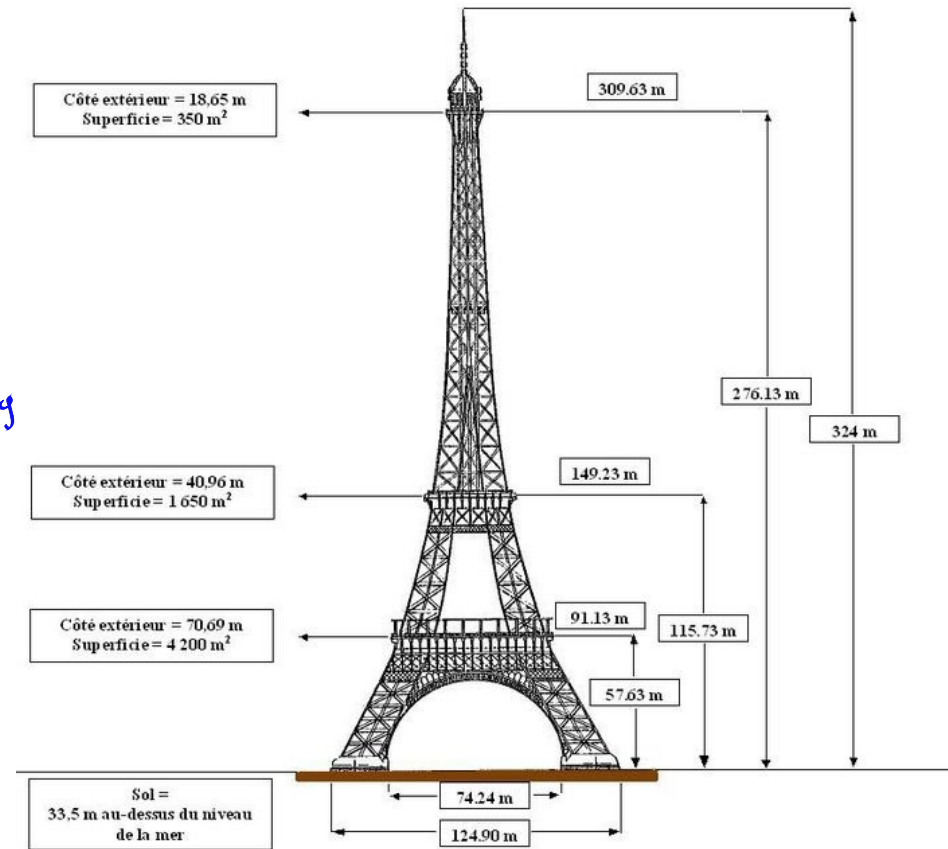
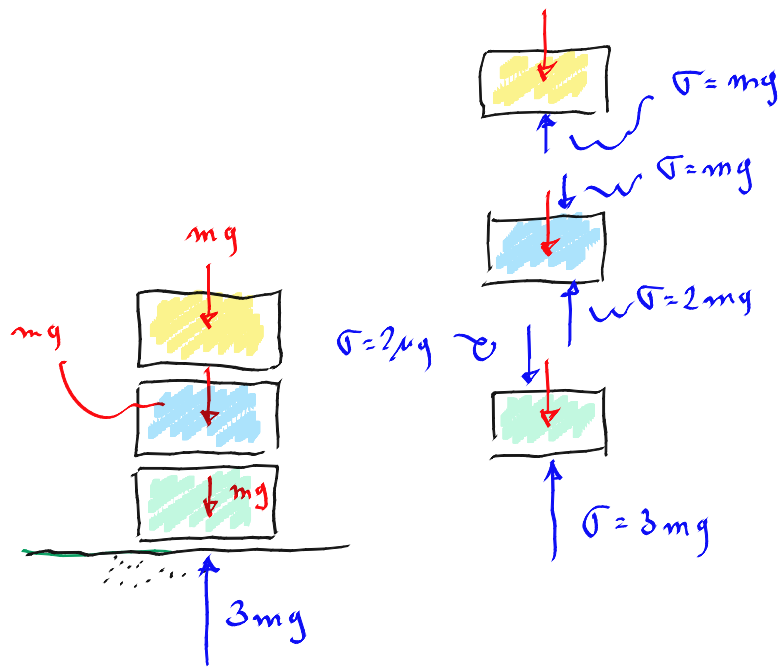
Boundary Value Problem:

$$\nabla \cdot \underline{\underline{\sigma}}(\underline{\underline{u}}) + \underline{\underline{f}} = 0 \quad \text{in } \Omega$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{S}} = \underline{\underline{g}} \quad \text{ON } \Gamma_2$$

$$\underline{\underline{u}} = 0 \quad \text{ON } \Gamma_0$$

Monsieur,
j'ai pas eu MMC, moi ?

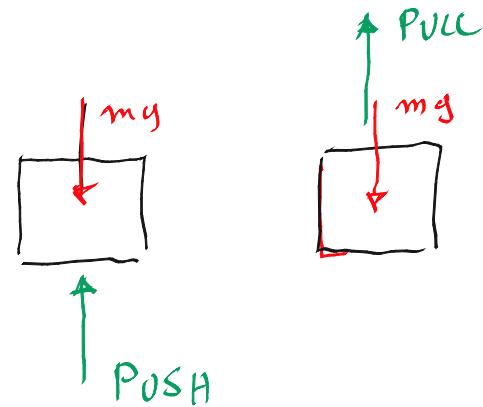
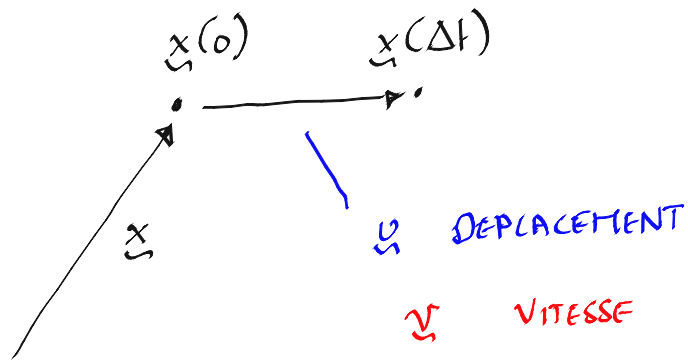


Ouuupps !

Mais, MMC, MSD et Flotte : quelle horreur !

Souviens-toi des anciens !
Le grand Newton !
L'inoubliable Roland !

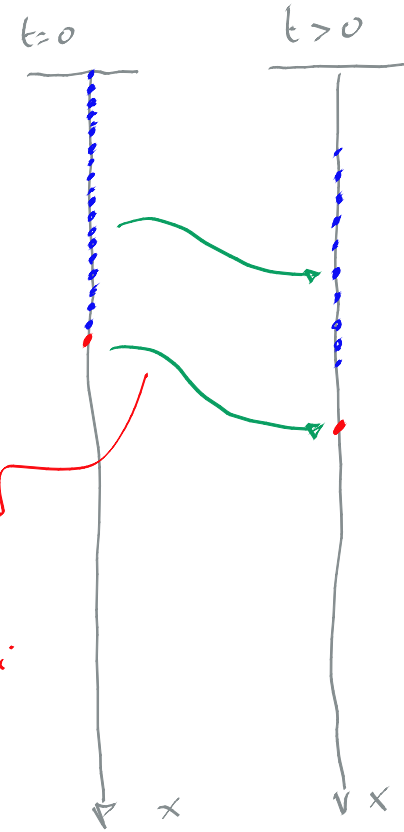
$$m \frac{d^2 x}{dt^2}(t) = \sum \underline{F}_i$$



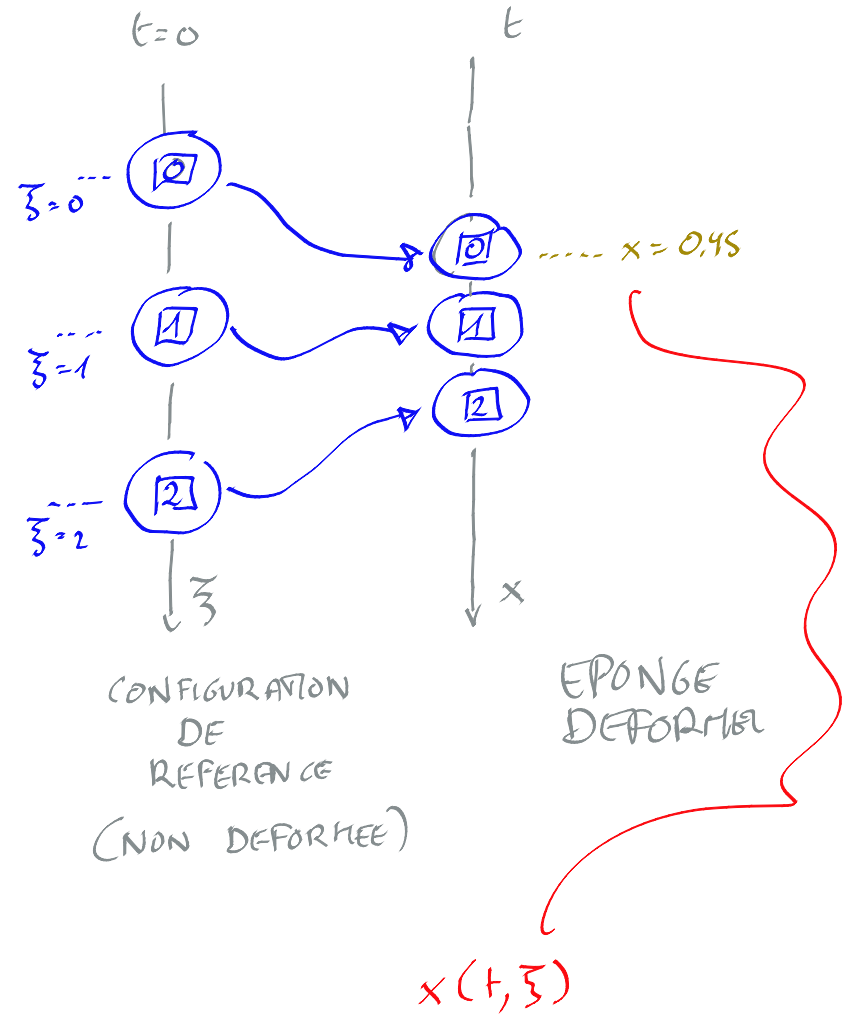
Un volume matériel !



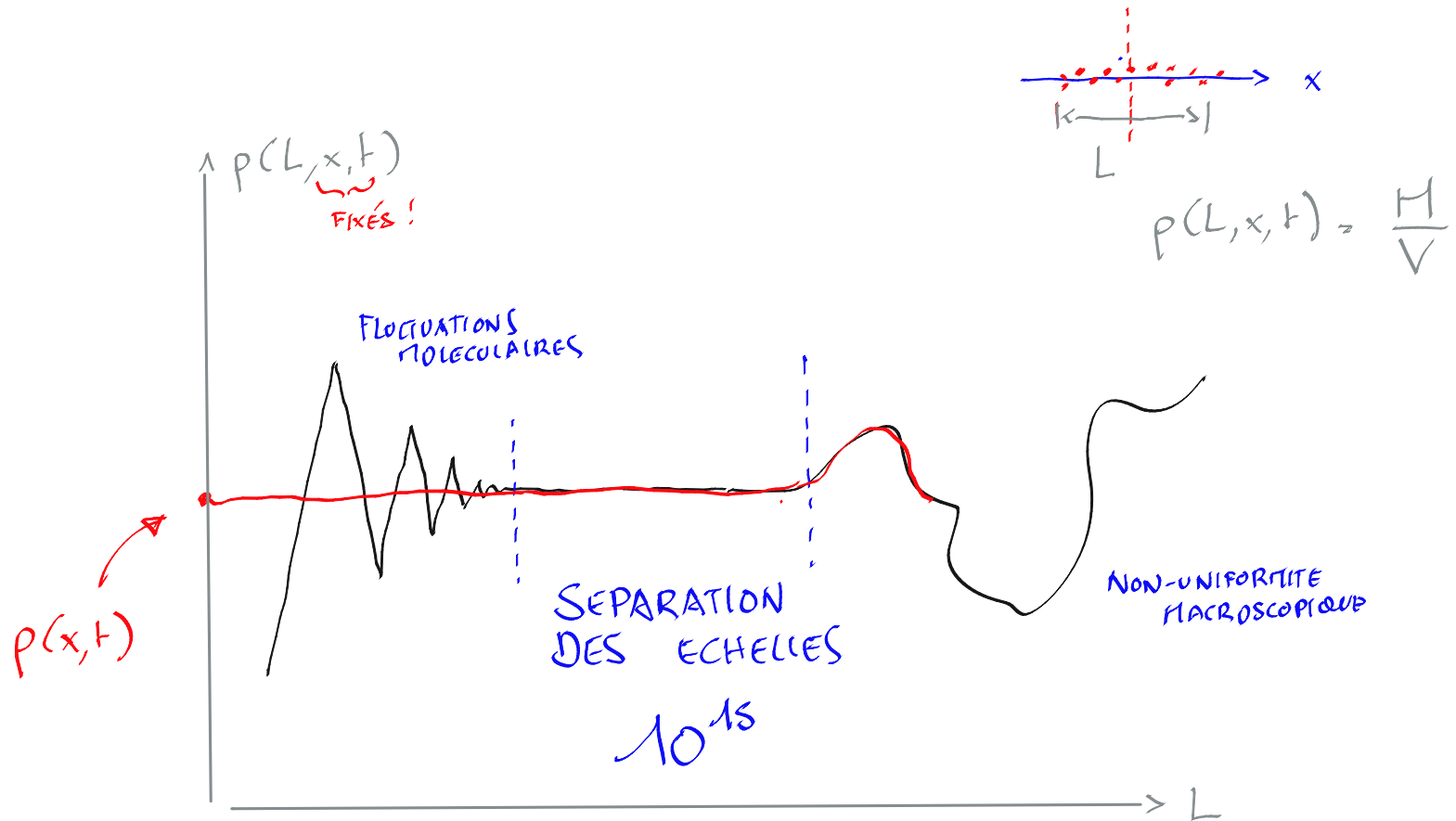
$$\underline{v}(x, t) = \sum v_i(x, t) \hat{e}_i$$



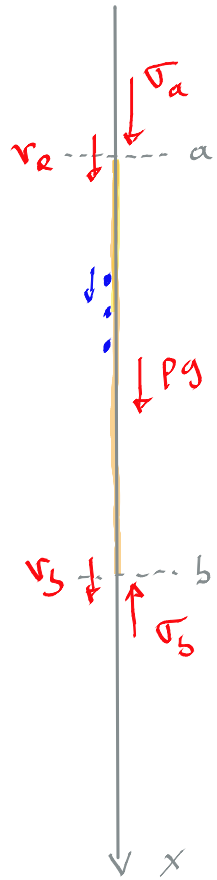
CONFIGURATION DE REFERENCE



La mécanique des milieux continus ! C'est une arnaque !



Bilan de quantité de mouvement



INTERVALLE
 a, b

ACCROISSEMENT
QUANTITE
DE MVT

= CE QUI ENTRE - CE QUI SORT + FORCES

$$\frac{d}{dt} \int_a^b \rho v \, dx = \underbrace{\rho_a v_a^2 - \rho_b v_b^2}_{- [\rho v^2]_a^b} + \int_a^b \rho g \, dx + \underbrace{(-\sigma_a) - (-\sigma_b)}_{[\sigma]_a^b}$$

$$= \int_a^b \frac{\partial}{\partial x} (\rho v^2) \, dx + \int_a^b \frac{\partial \sigma}{\partial x} \, dx$$

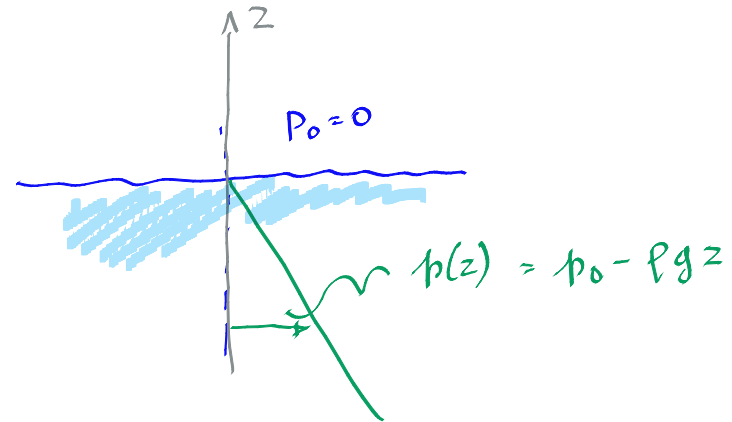
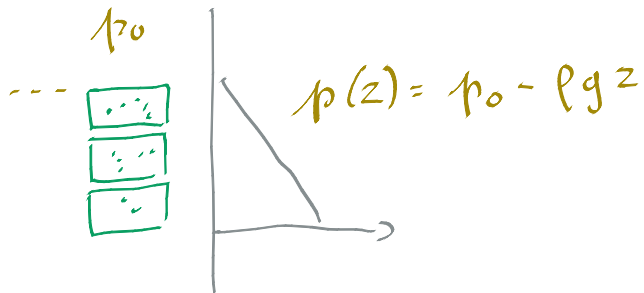
$$\underbrace{\frac{\partial}{\partial t} (\rho v) + \frac{\partial}{\partial x} (\rho v^2)}_{\rho \frac{Dv}{Dt}} = \rho g + \frac{\partial \sigma}{\partial x}$$

PROBLÈME STATIONNAIRE
PAS DE MOUVEMENT

Bilan de quantité de mouvement

$$\frac{Dv}{Dt} = 0$$

$$0 = \rho g + \frac{\partial \sigma}{\partial x}$$



PRESSION
HYDROSTATIQUE

Loi de Hooke !

So easy !

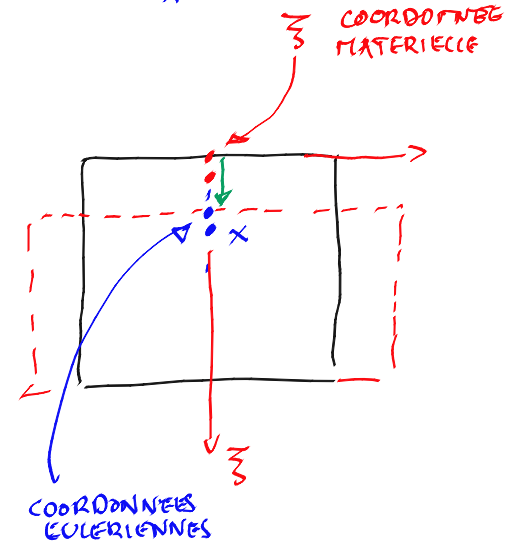
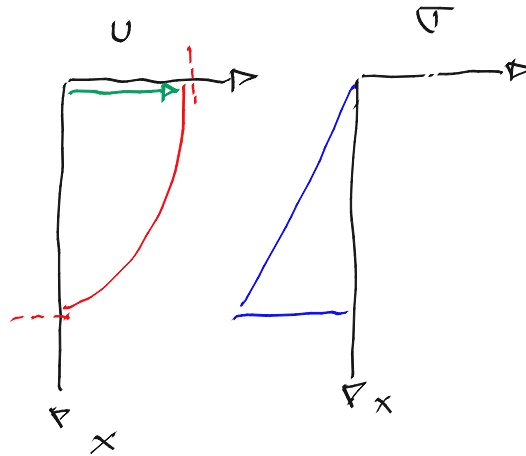
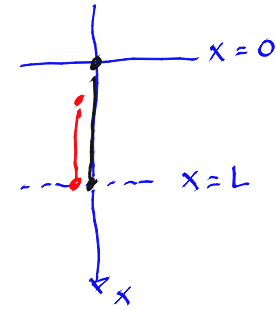
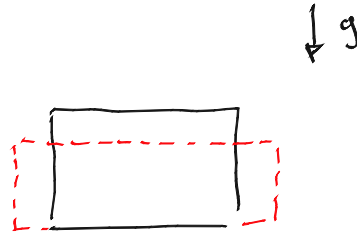
$$\frac{\partial \sigma}{\partial x} + \rho g = 0$$

$$\sigma = E \frac{du}{dx}$$

$$E \frac{d^2 u}{dx^2} = -\rho g$$

$$u(L) = 0$$

$$u'(0) = 0$$



Vitesse

\underline{v}



$\nabla_{\underline{v}}$

$$\frac{1}{2} (\nabla_{\underline{v}} + \nabla_{\underline{v}}^T)$$

Déplacement

\underline{u}



$\nabla_{\underline{u}}$

$$\frac{1}{2} (\nabla_{\underline{u}} + \nabla_{\underline{u}}^T)$$

$\underline{\underline{\epsilon}}$

$$\underline{\underline{D}} = 2\mu \underline{\underline{\epsilon}} + \lambda \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{S}}$$

$$\underline{\underline{\tau}} = \underline{\underline{d}} \underline{\underline{d}} - p \underline{\underline{S}}$$

Taux de déformation = gradient de vitesses

Déformation = gradient de déplacement

$$\sigma = E \frac{du}{dx}$$

$$\frac{du}{dx} \approx \frac{du}{d\xi}$$

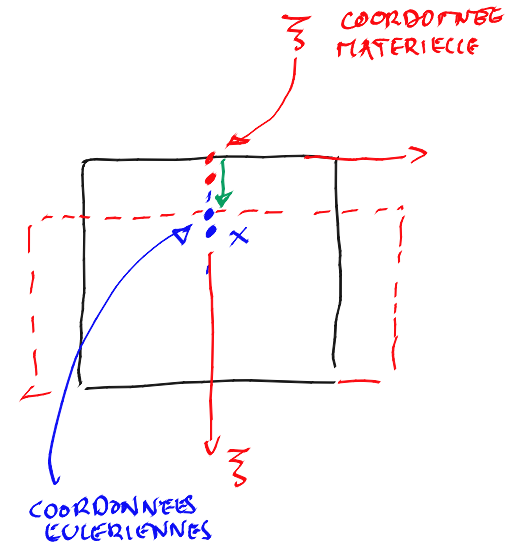
ON
FAIT

L'HYPOTHESE

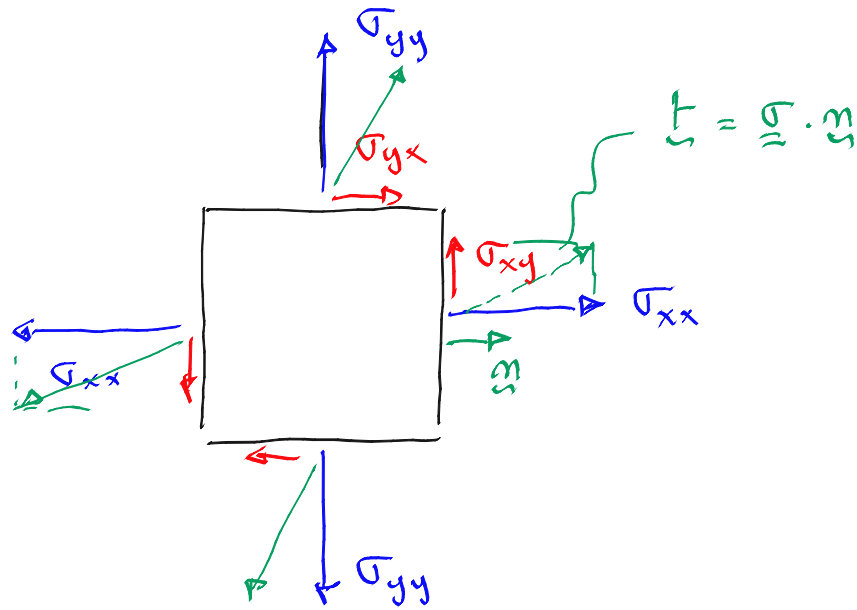
DE

PETITES

DEFORMATIONS

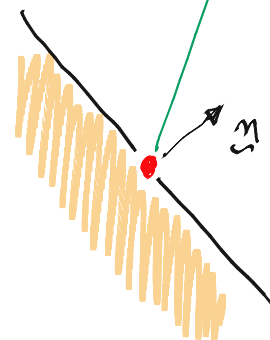


Grandes et petites déformations !
Cela peut devenir très compliqué !



$$\underline{\sigma}(\underline{x}) = \underline{\sigma} \cdot \underline{e}_3$$

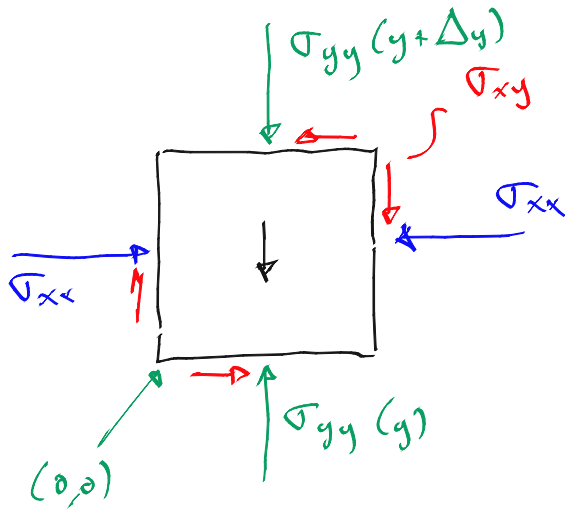
$$\underline{\sigma}(\underline{x}) = 119 \cdot \underline{e}_3$$



$$\underline{\sigma} = \begin{bmatrix} \sigma_{xx} & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} \end{bmatrix}$$

$$\underline{\sigma} = \underline{\sigma}^T$$

Tenseur
de contraintes



$$\rho g \Delta x \Delta y + \sigma_{yy}(y + \Delta y) \Delta x - \sigma_{yy}(y) \Delta x$$

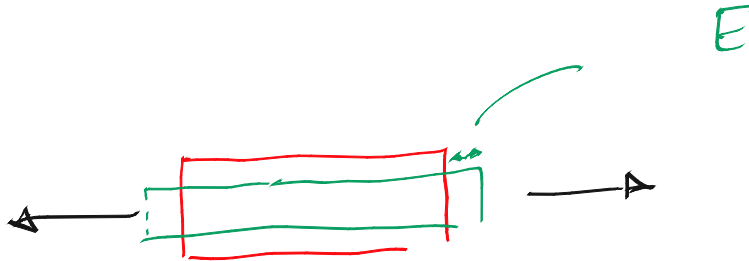
$$\begin{cases} \rho g + \frac{\partial}{\partial y}(\sigma_{yy}) + \frac{\partial}{\partial x}(\tau_{xy}) = 0 \\ \frac{\partial}{\partial x}(\sigma_{xx}) + \frac{\partial}{\partial y}(\tau_{xy}) = 0 \end{cases}$$

Bilan
de quantité de mouvement

$$\underline{\underline{\underline{\sigma}}}} = \underbrace{2\mu}_{\frac{E}{(1+\nu)}} \underline{\underline{\underline{\epsilon}}} + \lambda \operatorname{tr}(\underline{\underline{\underline{\epsilon}}}) \underline{\underline{\underline{1}}}$$

COEFFICIENTS DE LAMÉ

$\frac{E\nu}{(1+\nu)(1-2\nu)}$



Loi de Hooke

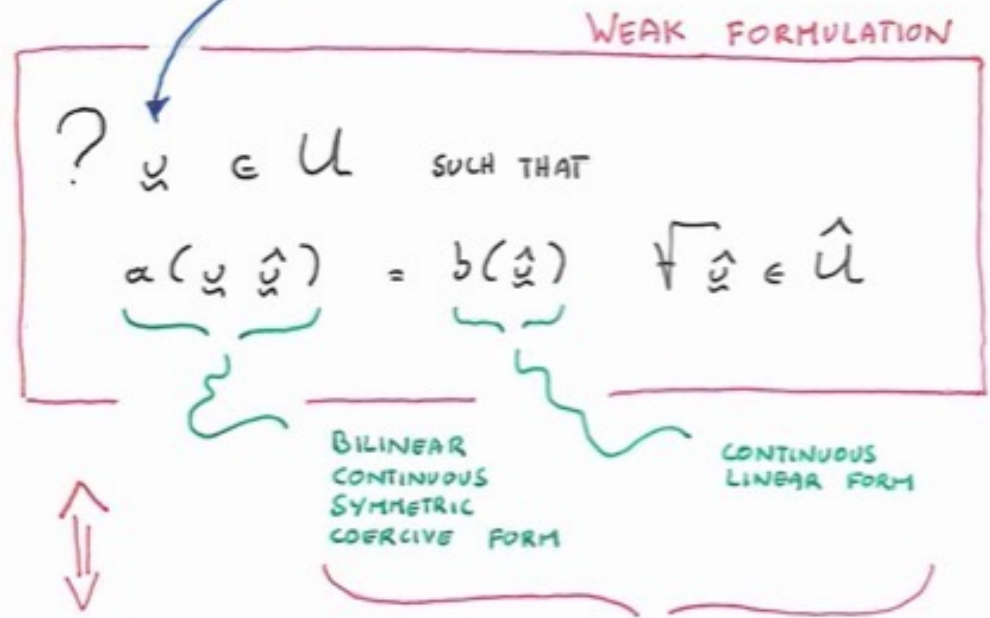
Coefficients de Lamé

Module de Young et coefficients de Poisson

ABSTRACT GENERIC ELLIPTIC PROBLEM

HEAT CONDUCTION
 LINEAR ELASTICITY
 SIMPLIFIED MODELS OF LINEAR ELASTICITY
 BEAM / SHELLS
 ROPE / MEMBRANE
 STOKES PROBLEM

Now,
 WE CONSIDER
 A VECTORIAL UNKNOWN FIELD !



? $\underline{u} \in \mathcal{U}$ SUCH THAT

$$\mathcal{J}(\underline{u}) = \min_{\underline{v} \in \mathcal{U}} \underbrace{\frac{1}{2} a(\underline{v}, \underline{v}) - b(\underline{v})}_{\mathcal{J}(\underline{v})}$$

ASSUMPTIONS
 REQUIRED
 TO HAVE AN ABSTRACT
 MINIMIZATION
 PROBLEM

MINIMIZATION
 PROBLEM

HEAT CONDUCTION

CONSERVATION LAW

$$-\nabla \cdot \underline{q} + f = 0$$

HEAT FLOW

CONSTITUTIVE LAW

$$\underline{q} = -k \nabla u$$

FOURIER

TEMPERATURE

? u SUCH THAT

$$\nabla \cdot (\overbrace{-\underline{q}(u)}^{k \nabla u}) + f = 0 \quad \text{in } \Omega$$

$$-\underline{q} \cdot \underline{n} = g \quad \text{ON } \Gamma_2$$

$$u = 0 \quad \text{ON } \Gamma_0$$

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$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{S}}$$

CONSERVATION LAWS

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}} = 0$$

$$\underline{\underline{\sigma}} = \underline{\underline{\sigma}}^T$$

DEFORMATION TENSOR

$$\underline{\underline{\epsilon}} \triangleq \frac{1}{2} (\nabla_{\underline{\underline{u}}} + (\nabla_{\underline{\underline{u}}})^T)$$

? $\underline{\underline{u}}$

$$\nabla \cdot \underline{\underline{\sigma}}(\underline{\underline{u}}) + \underline{\underline{f}} = 0 \quad \text{in } \Omega$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = \underline{\underline{g}} \quad \text{ON } \Gamma_2$$

$$\underline{\underline{u}} = 0 \quad \text{ON } \Gamma_0$$

STOKES PROBLEM

CONSERVATION LAWS

$$\cancel{(\underline{v} \cdot \nabla) \underline{v}} = \nabla \cdot \underline{\underline{\sigma}} + \underline{f}$$

$$\nabla \cdot \underline{v} = 0$$

CREEPING FLOW
 $Re \ll 0$

INCOMPRESSIBLE FLOW

CONSTITUTIVE LAW

NEWTONIAN FLUID

$$\underline{\underline{\sigma}} = 2\mu \underline{\underline{d}} - p \underline{\underline{I}}$$

RATE OF DEFORMATION TENSOR

$$\underline{\underline{d}} \triangleq \frac{1}{2} (\nabla \underline{v} + (\nabla \underline{v})^T)$$

? (\underline{v}, p)

$$\begin{cases} \nabla \cdot (\underline{\underline{\sigma}}(\underline{v}, p)) + \underline{f} = 0 \\ \nabla \cdot \underline{v} = 0 \end{cases} \quad \text{in } \Omega$$

$$\begin{aligned} \underline{\underline{\sigma}} \cdot \underline{e}_3 &= \underline{g} & \text{on } \sqrt{2} \\ \underline{v} &= \underline{0} & \text{on } \sqrt{2} \end{aligned}$$

Créer une géométrie !

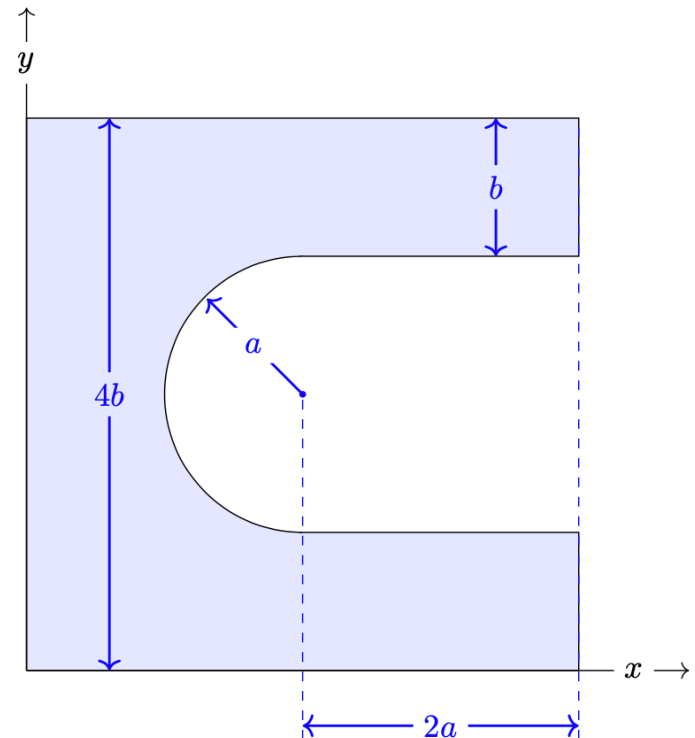
Construire le maillage !

Définir la géométrie et le maillage !

Le maillage et la géométrie sont définis comme suit :

```
double Lx = 1.0;
double Ly = 1.0;

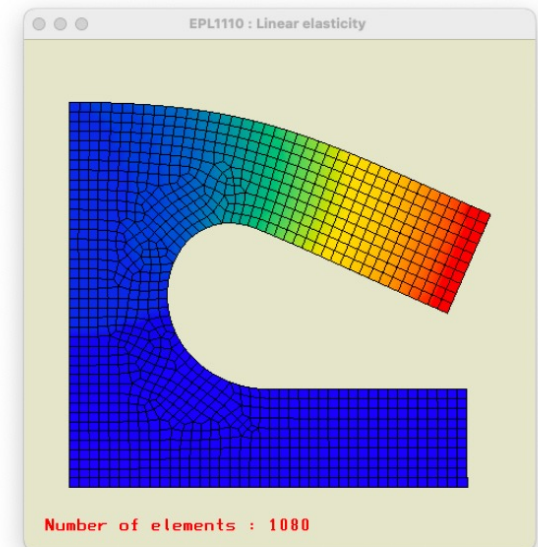
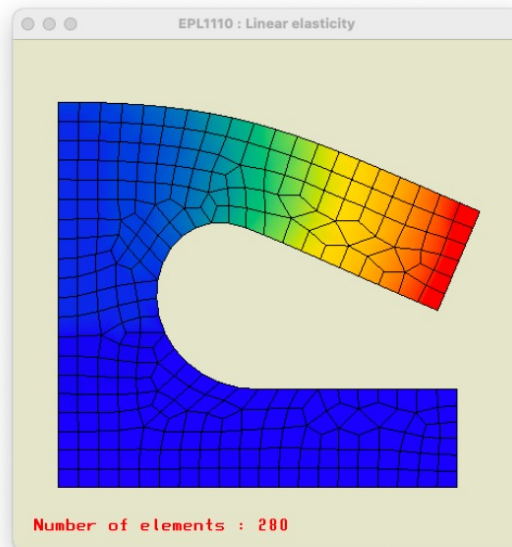
theGeometry->LxPlate = Lx;
theGeometry->LyPlate = Ly;
theGeometry->h = Lx * 0.05;
theGeometry->elementType = FEM_TRIANGLE;
```



Que va-t-on résoudre ?

Trouver $\mathbf{u}(\mathbf{x}) \in \mathcal{U}$ tel que

$$\underbrace{\langle \boldsymbol{\epsilon}(\hat{\mathbf{u}}) : \mathbf{C} : \boldsymbol{\epsilon}(\mathbf{u}) \rangle}_{a(\hat{\mathbf{u}}, \mathbf{u})} = \underbrace{\langle \hat{\mathbf{u}} f \rangle + \ll \hat{\mathbf{u}} g \gg_N}_{b(\hat{\mathbf{u}})}, \quad \forall \hat{\mathbf{u}} \in \hat{\mathcal{U}},$$



Et on minimise
toujours une fonctionnelle !

Trouver $\mathbf{u}(\mathbf{x}) \in \mathcal{U}$ tel que

$$\underbrace{\langle \boldsymbol{\epsilon}(\hat{\mathbf{u}}) : \mathbf{C} : \boldsymbol{\epsilon}(\mathbf{u}) \rangle}_{a(\hat{\mathbf{u}}, \mathbf{u})} = \underbrace{\langle \hat{\mathbf{u}}f \rangle + \ll \hat{\mathbf{u}}g \gg_N}_{b(\hat{\mathbf{u}})}, \quad \forall \hat{\mathbf{u}} \in \hat{\mathcal{U}},$$

$$J(\mathbf{v}) = \underbrace{\frac{1}{2} \langle \boldsymbol{\epsilon}(\mathbf{v}) : \mathbf{C} : \boldsymbol{\epsilon}(\mathbf{v}) \rangle}_{\frac{1}{2}a(\mathbf{v}, \mathbf{v})} - \underbrace{\langle \mathbf{v}f \rangle + \ll \mathbf{v}g \gg_N}_{b(\mathbf{v})},$$

En deux dimensions !

$$\begin{bmatrix} \sigma_{xx} & \sigma_{xy} \\ \sigma_{xy} & \sigma_{yy} \end{bmatrix} = \begin{bmatrix} A\epsilon_{xx} + B\epsilon_{yy} & 2C\epsilon_{xy} \\ 2C\epsilon_{xy} & A\epsilon_{yy} + B\epsilon_{xx} \end{bmatrix}$$

Acier

$$E = 2.11 \cdot 10^{11} \text{ [N/m}^2\text{]}$$

$$\nu = 0.3$$

$$\rho = 7.85 \cdot 10^3 \text{ [kg/m}^3\text{]}$$

Déformations planes

Tensions planes

$$A = \frac{E(1 - \nu)}{(1 + \nu)(1 - 2\nu)}$$

$$A = \frac{E}{(1 - \nu^2)}$$

$$B = \frac{E\nu}{(1 + \nu)(1 - 2\nu)}$$

$$B = \frac{E\nu}{(1 - \nu^2)}$$

$$C = \frac{E}{2(1 + \nu)}$$

Déformations planes...

$$\sigma_{xx} = \frac{E}{(1+\nu)(1-2\nu)} ((1-\nu)\epsilon_{xx} + \nu\epsilon_{yy}),$$

$$\sigma_{yy} = \frac{E}{(1+\nu)(1-2\nu)} (\nu\epsilon_{xx} + (1-\nu)\epsilon_{yy}),$$

$$\sigma_{zz} = \frac{E\nu}{(1+\nu)(1-2\nu)} (\epsilon_{xx} + \epsilon_{yy}),$$

$$\sigma_{xy} = \frac{E}{(1+\nu)} \epsilon_{xy},$$

$$\sigma_{xx} = \frac{E}{(1-\nu^2)} (\epsilon_{xx} + \nu\epsilon_{yy}),$$

$$\sigma_{yy} = \frac{E}{(1-\nu^2)} (\nu\epsilon_{xx} + \epsilon_{yy}),$$

$$\sigma_{xy} = \frac{E}{(1+\nu)} \epsilon_{xy}.$$

...tensions planes

La fonctionnelle à minimiser !

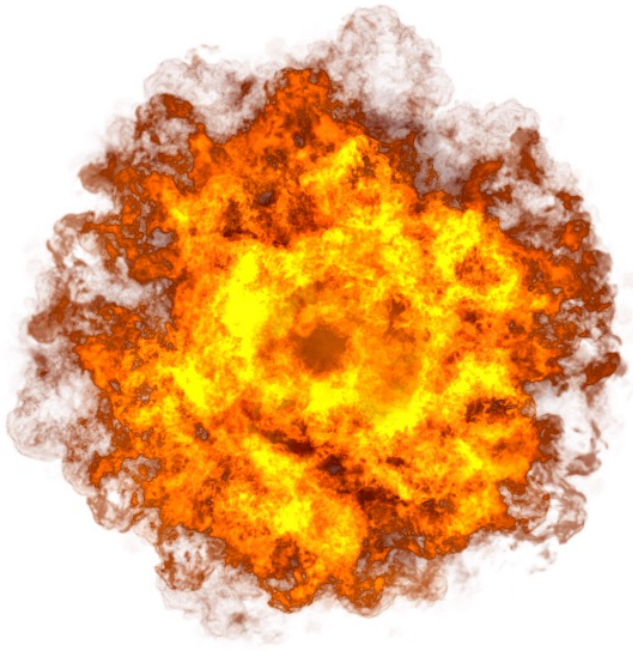
$$\begin{aligned}\frac{1}{2} a(\mathbf{u}, \mathbf{u}) &= \frac{1}{2} \langle \boldsymbol{\epsilon}(\mathbf{u}) : \boldsymbol{\sigma}(\mathbf{u}) \rangle, \\ &= \frac{1}{2} \langle \begin{bmatrix} \epsilon_{xx} & \epsilon_{xy} \\ \epsilon_{xy} & \epsilon_{yy} \end{bmatrix} : \begin{bmatrix} A\epsilon_{xx} + B\epsilon_{yy} & 2C\epsilon_{xy} \\ 2C\epsilon_{xy} & A\epsilon_{yy} + B\epsilon_{xx} \end{bmatrix} \rangle, \\ &= \frac{1}{2} \langle \begin{bmatrix} \epsilon_{xx} & \epsilon_{yy} & 2\epsilon_{xy} \end{bmatrix} \cdot \begin{bmatrix} A & B & 0 \\ B & A & 0 \\ 0 & 0 & C \end{bmatrix} \cdot \begin{bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ 2\epsilon_{xy} \end{bmatrix} \rangle.\end{aligned}$$

Matrice et vecteur locaux à construire !

$$\mathbf{A}_{ij} = \left[\begin{array}{c|c} \langle \tau_{i,x} A \tau_{j,x} \rangle + \langle \tau_{i,y} C \tau_{j,y} \rangle & \langle \tau_{i,x} B \tau_{j,y} \rangle + \langle \tau_{i,y} C \tau_{j,x} \rangle \\ \hline \langle \tau_{i,y} B \tau_{j,x} \rangle + \langle \tau_{i,x} C \tau_{j,y} \rangle & \langle \tau_{i,y} A \tau_{j,y} \rangle + \langle \tau_{i,x} C \tau_{j,x} \rangle \end{array} \right],$$

$$\mathbf{B}_i = \left[\begin{array}{c} \langle \tau_i f_x \rangle + \ll \tau_i g_x \gg \\ \langle \tau_i f_y \rangle + \ll \tau_i g_y \gg \end{array} \right].$$

Et on soumet au serveur...



```
=====
Linear elasticity problem
Young modulus E = 2.1100000e+11 [N/m2]
Poisson's ratio nu = 3.0000000e-01 [-]
Density rho = 7.8500000e+03 [kg/m3]
Gravity g = 9.8100000e+00 [m/s2]
Planar strains formulation
Boundary conditions :
    Symmetry : imposing 0.00e+00 as the horizontal displacement
    Bottom : imposing 0.00e+00 as the vertical displacement
=====
```

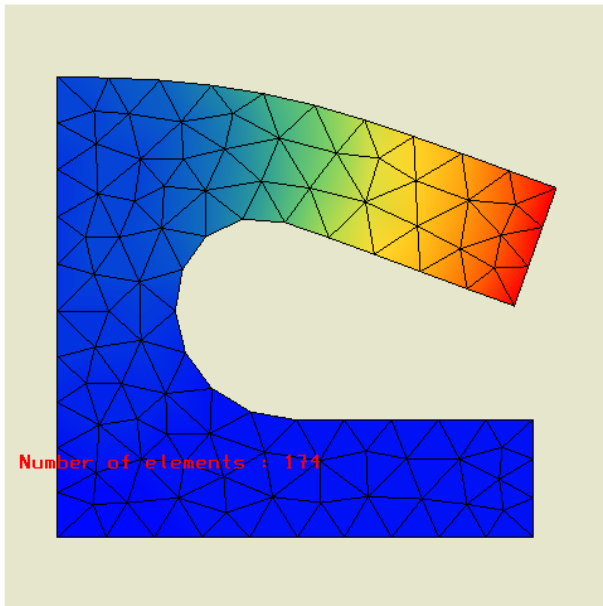
```
Pivot index 1 Pivot value 0.000000e+00
```

```
-----
Error in src/fem.c at line 555 :
Cannot eliminate with such a pivot
```

```
----- Yek Yek !!
```

```
La compilation ou l'execution du test 0 a plante :-(
```

Et après avoir fait le devoir...



Diagnostic du serveur

```
***** TEST 0 *****  
gcc -o /tmp/work/exe src/glfem.c src/main.c src/fem.c src/mesaglfem.c homework.c -I src -lm -lrt -Wall -g -Wno-  
** COMPILATION SUCCEEDED **  
  
/usr/bin/time -o /work/outputStat.txt -f %R %Z %e %M sudo -ustudent ./exe  
** RUN SUCCEEDED **  
Wall-clock time : 0.24s (limit 40s)  
Approximated maximum memory usage : 112.98Mb  
Approximated total memory allocation : 28.8203Mb  
***** RUN OUTPUT *****  
Info : Meshing 1D...  
Info : [ 0%] Meshing curve 1 (Line)  
Info : [ 20%] Meshing curve 2 (Line)  
Info : [ 30%] Meshing curve 3 (Line)
```


PROBLÈME ELLIPTIQUE



ELEMENTS FINIS *



* "USUAL FEM"
CLASSICAL
GALERKIN
FORMULATION

ABSTRACT GENERIC ELLIPTIC PROBLEM

HEAT CONDUCTION
 LINEAR ELASTICITY
 SIMPLIFIED MODELS OF LINEAR ELASTICITY
 BEAM / SHELLS
 ROPE / MEMBRANE
 STOKES PROBLEM

Now,
 WE CONSIDER
 A VECTORIAL UNKNOWN FIELD !

WEAK FORMULATION

? $\underline{u} \in \mathcal{U}$ SUCH THAT

$$\underbrace{a(\underline{u}, \hat{\underline{u}})}_{\substack{\text{BILINEAR} \\ \text{CONTINUOUS} \\ \text{SYMMETRIC} \\ \text{COERCIVE FORM}}} = \underbrace{b(\hat{\underline{u}})}_{\text{CONTINUOUS LINEAR FORM}} \quad \forall \hat{\underline{u}} \in \hat{\mathcal{U}}$$

↕

ASSUMPTIONS
 REQUIRED
 TO HAVE AN ABSTRACT
 MINIMIZATION
 PROBLEM

? $\underline{u} \in \mathcal{U}$ SUCH THAT

$$\mathcal{J}(\underline{u}) = \min_{\underline{v} \in \mathcal{U}} \underbrace{\frac{1}{2} a(\underline{v}, \underline{v}) - b(\underline{v})}_{\mathcal{J}(\underline{v})}$$

MINIMIZATION
 PROBLEM

HEAT CONDUCTION

CONSERVATION LAW

$$-\nabla \cdot \underline{q} + f = 0$$

HEAT FLOW

CONSTITUTIVE LAW

$$\underline{q} = -k \nabla u$$

FOURIER

TEMPERATURE

? u SUCH THAT

$$\nabla \cdot (\overbrace{-\underline{q}(u)}^{k \nabla u}) + f = 0 \quad \text{in } \Omega$$

ON Γ_2

ON Γ_0

$$-\underline{q} \cdot \underline{n} = g$$

$$u = 0$$

3D LINEAR ISOTROPIC ELASTICITY



HOOKE ELASTIC BODY
SMALL DEF.

CONSTITUTIVE LAW

$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{1}}$$

CONSERVATION LAWS

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}} = 0$$

$$\underline{\underline{\sigma}} = \underline{\underline{\sigma}}^T$$

DEFORMATION TENSOR

$$\underline{\underline{\epsilon}} \triangleq \frac{1}{2} (\nabla_{\underline{\underline{u}}} + (\nabla_{\underline{\underline{u}}})^T)$$

Boundary Value Problem:

$$\nabla \cdot \underline{\underline{\sigma}}(\underline{\underline{u}}) + \underline{\underline{f}} = 0 \quad \text{in } \Omega$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = \underline{\underline{g}} \quad \text{ON } \Gamma_2$$

$$\underline{\underline{u}} = 0 \quad \text{ON } \Gamma_0$$

STOKES PROBLEM

CONSERVATION LAWS

$$\cancel{(\underline{v} \cdot \nabla) \underline{v}} = \nabla \cdot \underline{\underline{g}} + \underline{f}$$

$$\nabla \cdot \underline{v} = 0$$

CREEPING FLOW
 $Re \ll 0$

INCOMPRESSIBLE FLOW

CONSTITUTIVE LAW

NEWTONIAN FLUID

$$\underline{\underline{g}} = 2\mu \underline{\underline{d}} \rightarrow \underline{\underline{s}}$$

RATE OF DEFORMATION TENSOR

$$\underline{\underline{d}} \triangleq \frac{1}{2} (\nabla \underline{v} + (\nabla \underline{v})^T)$$

? (\underline{v}, p)

$$\left\{ \begin{array}{l} \nabla \cdot (\underline{\underline{g}}(\underline{v}, p)) + \underline{f} = 0 \\ \nabla \cdot \underline{v} = 0 \end{array} \right. \quad \text{in } \Omega$$

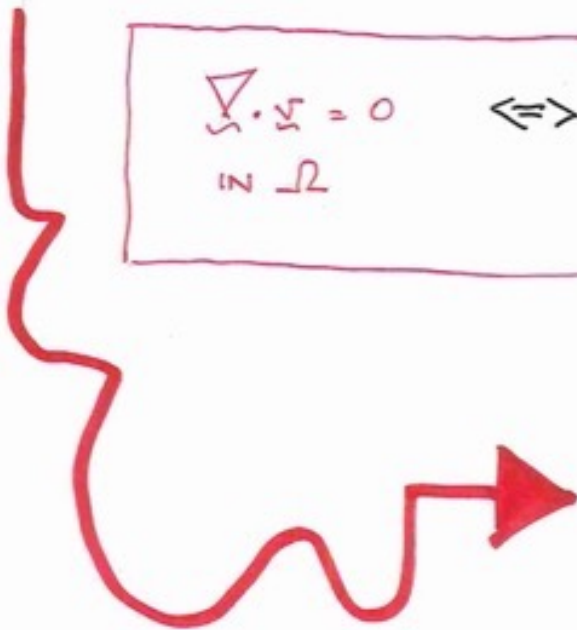
$$\underline{\underline{g}} \cdot \underline{e}_3 = \underline{g} \quad \text{on } \sqrt{2}$$

$$\underline{v} = 0 \quad \text{on } \sqrt{0}$$

CALCULUS

$$\nabla \cdot \underline{v} = 0 \quad \Leftrightarrow \quad \exists \psi \quad \text{SUCH THAT}$$

$$\underline{v} = \nabla \times \underline{\psi}$$



STREAM FUNCTION

? $\underline{\psi}$

$$\nabla^4 \underline{\psi} + \nabla \times \underline{f} = 0 \quad \text{in } \Omega$$

+ SUITABLE BOUNDARY CONDITIONS

3D LINEAR ISOTROPIC ELASTICITY



HOOKE ELASTIC BODY
SMALL DEF.

CONSTITUTIVE LAW

$$\underline{\underline{\sigma}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} \text{tr}(\underline{\underline{\epsilon}}) \underline{\underline{S}}$$

CONSERVATION LAWS

$$\nabla \cdot \underline{\underline{\sigma}} + \underline{\underline{f}} = 0$$

$$\underline{\underline{\sigma}} = \underline{\underline{\sigma}}^T$$

DEFORMATION TENSOR

$$\underline{\underline{\epsilon}} \triangleq \frac{1}{2} (\nabla_{\underline{\underline{u}}} + (\nabla_{\underline{\underline{u}}})^T)$$

Boundary Value Problem:

$$\nabla \cdot \underline{\underline{\sigma}}(\underline{\underline{u}}) + \underline{\underline{f}} = 0 \quad \text{in } \Omega$$

$$\underline{\underline{\sigma}} \cdot \underline{\underline{S}} = \underline{\underline{g}} \quad \text{ON } \Gamma_2$$

$$\underline{\underline{u}} = 0 \quad \text{ON } \Gamma_0$$

$$\underline{\underline{D}} = \underbrace{\frac{E}{(1+\nu)}}_{2\mu \text{ SHEAR MODULUS}} \underline{\underline{\epsilon}} + \underbrace{\frac{E\nu}{(1+\nu)(1-2\nu)}}_{\lambda} t_n(\underline{\underline{\epsilon}}) \underline{\underline{S}}$$

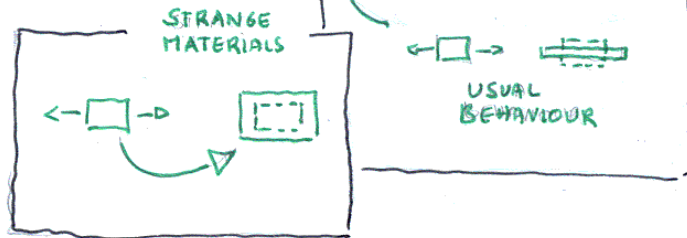
POISSON'S COEFFICIENT

$$\nu \in]-1$$

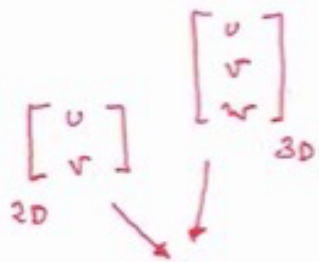
$$\frac{1}{2} [$$

INCOMPRESSIBLE MATERIAL

$$\lambda \rightarrow \infty \text{ IF } \nu \rightarrow \frac{1}{2}$$



ABSTRACT GENERIC DISCRETE FORMULATION



$$\underbrace{u(x)}_{\in U}$$

$$u^h(x) = \sum_{i=1}^m \tau_i(x)$$

$$\in U^h \subset U$$

$$\dim(U^h) = 2m$$



2D PROBLEMS

$$\dots = 3m$$



3D PROBLEMS

? u^h SUCH THAT

$$a(u^h, \hat{u}^h) = b(\hat{u}^h)$$

$$\forall \hat{u}^h \in \hat{U}^h$$

DISCRETE
FORMULATION

$$U^h = \text{SPAN} \left\{ \underbrace{\begin{bmatrix} \tau_1 \\ 0 \end{bmatrix} \begin{bmatrix} \tau_2 \\ 0 \end{bmatrix} \dots \begin{bmatrix} \tau_m \\ 0 \end{bmatrix}}_m, \underbrace{\begin{bmatrix} 0 \\ \tau_1 \end{bmatrix} \dots \begin{bmatrix} 0 \\ \tau_m \end{bmatrix}}_m \right\}$$

TIP #1

$$\langle \hat{u} \cdot (\nabla \cdot \underline{g}(u)) \rangle = \langle \nabla \cdot (\hat{u} \cdot \underline{g}(u)) \rangle - \langle \nabla \hat{u} : \underline{g}(u) \rangle$$

$\epsilon \hat{u}$ ϵu

$\equiv a(\hat{u}, u)$

TIP #2

$$= \langle \nabla \cdot \hat{u} \cdot \underline{g}(u) \rangle$$

TIP #3

$$= \langle \nabla \cdot \left(\underbrace{\hat{u}}_{\epsilon \hat{u}} \cdot \underbrace{\underline{g}(u)}_{\epsilon u} \right) \rangle$$

USUAL CALCULUS !

IS $\alpha(\hat{u}, u)$
SYMMETRIC?

$$\mathbb{D} = \mathbb{C} : \underline{\underline{\epsilon}}$$

$$\langle \nabla \hat{u} : \mathbb{D}(u) \rangle = \langle \underbrace{\left(\frac{\nabla \hat{u} + (\nabla \hat{u})^T}{2} \right)}_{\text{SYM}} + \underbrace{\left(\frac{\nabla \hat{u} - (\nabla \hat{u})^T}{2} \right)}_{\text{ANTI-SYM}} : \underbrace{\mathbb{D}(u)}_{\text{SYM}} \rangle$$



$$= \langle \underline{\underline{\epsilon}}(\hat{u}) : \mathbb{D}(u) \rangle$$



$$\mathbb{C} : \underline{\underline{\epsilon}}(u)$$

GENERALIZED HOOKE'S LAW

$$= \langle \underline{\underline{\epsilon}}(\hat{u}) : \mathbb{C} : \underline{\underline{\epsilon}}(u) \rangle$$

WEAK FORMULATION

? $u \in U$ SUCH THAT

$$\underbrace{\langle \underline{\underline{\underline{\varepsilon}}}}_{a(\hat{u}, u)} : \underbrace{\underline{\underline{\underline{\varepsilon}}}}_{\varepsilon(u)} \rangle = \underbrace{\langle \hat{u} \cdot f \rangle}_{b(\hat{u})} + \underbrace{\langle \hat{u} \cdot g \rangle}_N$$

$\forall \hat{u} \in \hat{U}$

FOR SUITABLE ONLY \subseteq !

? $u \in U$ SUCH THAT

$$J(u) = \min_{v \in U} \underbrace{\frac{1}{2} a(u, v)}_{J(v)} - b(v)$$

MINIMIZATION PROBLEM

$$\frac{1}{2} \alpha(\underline{v}, \underline{v}) = \left\langle \frac{1}{2} \lambda \underline{\underline{\epsilon}}(\underline{v}) : \underline{\underline{\epsilon}}(\underline{v}) + \mu \underline{\underline{\epsilon}}(\underline{v}) : \underline{\underline{\epsilon}}(\underline{v}) \right\rangle$$

ENERGY
OF
DEFORMATION

MUST BE
A QUADRATIC POSITIVE FORM
IN ORDER TO OBTAIN A MINIMIZATION
PROBLEM

$$\begin{array}{l} \mu > 0 \\ \frac{3}{2} \lambda + \mu > 0 \end{array}$$

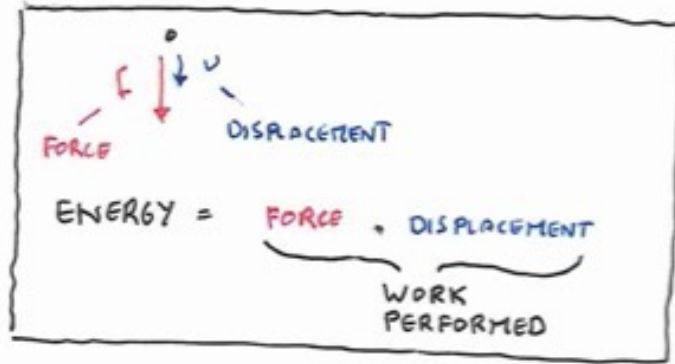
$$(\epsilon_{12} \neq 0)$$

$$(\epsilon_{11} = \epsilon_{22} = \epsilon_{33} \neq 0)$$

ADMISSIBLE VALUES
FOR LAMÉ COEFFICIENTS

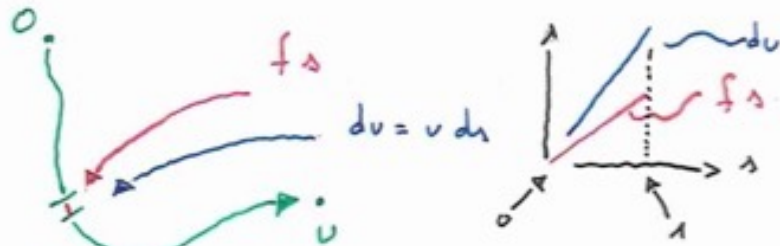
"DEFORMATION ENERGY"

WHAT IS IT?



ENERGY OF DEFORMATION

= WORK PERFORMED TO DEFORM THE STRUCTURE AND IS STORED INSIDE. SUCH AN ENERGY IS RECOVERED WHEN EXTERNAL FORCES ARE REMOVED



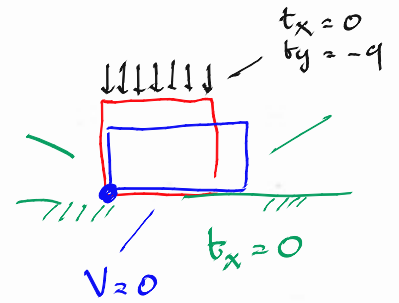
VERY VERY SLOW LOADING
→ QUASI-STATIC APPROACH
• IRREVERSIBILITY } ARE NEGLECTED!
• DISSIPATION

WORK PERFORMED

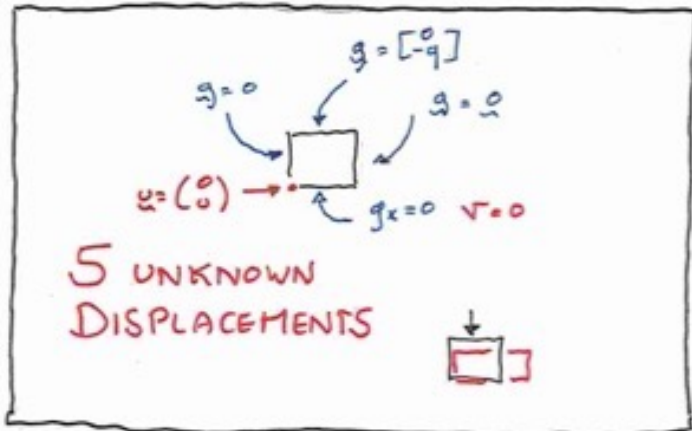
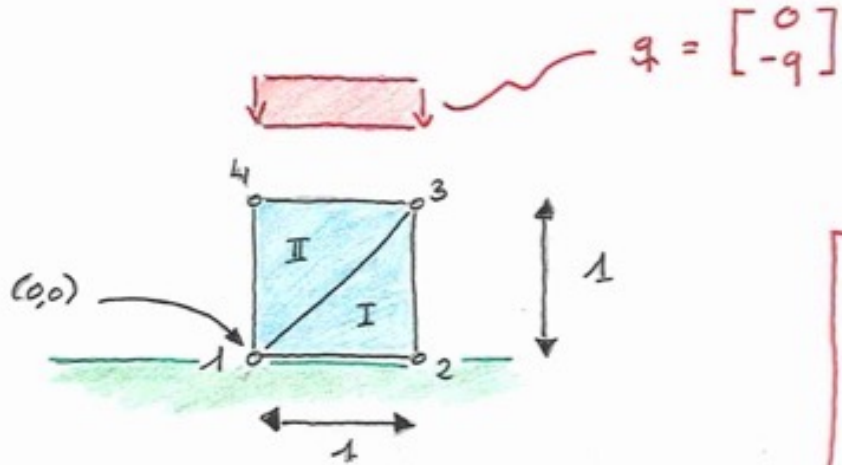
$$\begin{aligned}
 &= \int_0^u f \, du \\
 &= \int_0^1 f u \, ds \\
 &= f u \int_0^1 ds \\
 &= \frac{1}{2} f u !
 \end{aligned}$$

NUMERICAL EXAMPLE

$$t_x = t_y = 0$$



$$\xi = \begin{bmatrix} u \\ v \end{bmatrix}$$



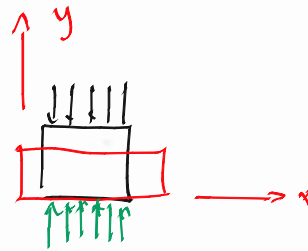
$$? \underline{u}^h = \sum_{i=1}^m \underline{U}_i \tau_i$$

$$\langle \underline{\underline{\epsilon}}(\underline{u}^h) : \underline{\underline{\sigma}}(\underline{u}^h) \rangle = \langle \underline{f} \cdot \underline{u}^h \rangle + \langle \underline{g} \cdot \underline{u}^h \rangle_N$$

$$\begin{bmatrix} \tau_i \\ 0 \end{bmatrix} \begin{bmatrix} 0 \\ \tau_i \end{bmatrix}$$

2m TEST FUNCTIONS

ANALYTICAL SOLUTION

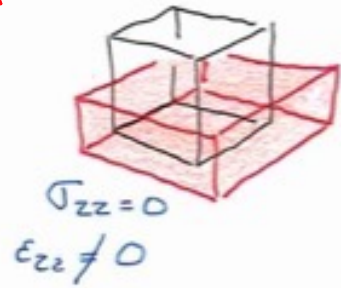


PLANAR STRESSES

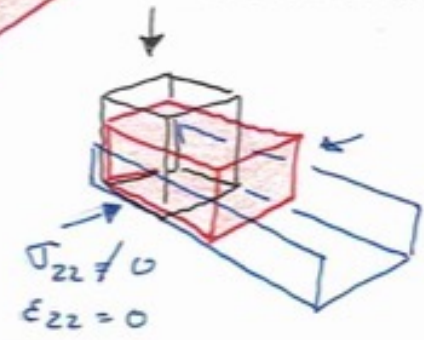
$$u = \frac{q}{E} \begin{bmatrix} ux \\ -y \end{bmatrix}$$

$$\epsilon = \frac{q}{E} \begin{bmatrix} \nu & 0 \\ 0 & -1 \end{bmatrix}$$

PLANAR STRESSES



PLANAR DEFORMATIONS



$$\Delta u = 0$$

= 0!

$$\sigma_{xx} = \frac{E}{(1-\nu^2)} \left(\underbrace{\epsilon_{xx}}_{\frac{q\nu}{E}} + \nu \underbrace{\epsilon_{yy}}_{-\frac{q}{E}} \right) = 0$$

$$\sigma_{yy} = \frac{E}{(1-\nu^2)} \left(\nu \underbrace{\epsilon_{xx}}_{\frac{q}{E}} - \underbrace{\epsilon_{yy}}_{(1-\nu^2)} \right) = -q!$$

$$\begin{bmatrix} 1 \\ 0 \end{bmatrix} \text{ OR } \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

$$\langle \begin{bmatrix} 1 \\ 0 \end{bmatrix} : \begin{bmatrix} 1 \\ 0 \end{bmatrix} \rangle$$

$$\sum_j U_j \begin{bmatrix} 1 \\ 0 \end{bmatrix} + V_j \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

$$\sum_j A_{ij} \cdot U_j$$

$$\sum_j \begin{bmatrix} U_j \\ V_j \end{bmatrix} \tau_j = \sum_j U_j \begin{bmatrix} \tau_j \\ 0 \end{bmatrix} + V_j \begin{bmatrix} 0 \\ \tau_j \end{bmatrix}$$

$$\sum_j \begin{bmatrix} \langle \begin{bmatrix} 1 \\ 0 \end{bmatrix} : \begin{bmatrix} 1 \\ 0 \end{bmatrix} \rangle & \langle \begin{bmatrix} 1 \\ 0 \end{bmatrix} : \begin{bmatrix} 0 \\ 1 \end{bmatrix} \rangle \\ \langle \begin{bmatrix} 0 \\ 1 \end{bmatrix} : \begin{bmatrix} 1 \\ 0 \end{bmatrix} \rangle & \langle \begin{bmatrix} 0 \\ 1 \end{bmatrix} : \begin{bmatrix} 0 \\ 1 \end{bmatrix} \rangle \end{bmatrix} \cdot \begin{bmatrix} U_j \\ V_j \end{bmatrix}$$

DISCRETE OPERATOR

$$\underline{\underline{\sigma}} \begin{pmatrix} \tau_{iz} \\ 0 \end{pmatrix} = \begin{bmatrix} \tau_{iz,x} & \tau_{iz,y}/2 \\ \tau_{iz,y}/2 & 0 \end{bmatrix}$$

$$\underline{\underline{\sigma}} \begin{pmatrix} 0 \\ \tau_{iz} \end{pmatrix} = \begin{bmatrix} 0 & \tau_{iz,x}/2 \\ \tau_{iz,x}/2 & \tau_{iz,y} \end{bmatrix}$$

$$\underline{\underline{G}} \begin{pmatrix} \underline{\underline{\epsilon}} \end{pmatrix} = \begin{bmatrix} A \epsilon_{xx} + B \epsilon_{yy} & 2C \epsilon_{xy} \\ 2C \epsilon_{xy} & A \epsilon_{yy} + B \epsilon_{xx} \end{bmatrix}$$

HOW
TO
CALCULATE
IT ?

$\frac{E(1-\nu)}{(1+\nu)(1-2\nu)}$	$\frac{E}{(1-\nu^2)}$
$\frac{E\nu}{(1+\nu)(1-2\nu)}$	$\frac{E\nu}{(1-\nu^2)}$
	$\frac{E}{2(1+\nu)}$

PLANAR
DEFORMATIONS

PLANAR
STRESSES

A

B

C

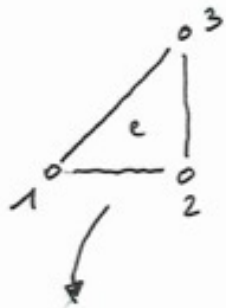
$$\underline{\underline{\sigma}} : \underline{\underline{\epsilon}} = \epsilon_{xx} (A \epsilon_{xx} + B \epsilon_{yy}) + 4C \epsilon_{xy} \epsilon_{xy} + \epsilon_{yy} (A \epsilon_{yy} + B \epsilon_{xx})$$

$$\underline{\underline{A}}_y = \begin{bmatrix} \langle \tau_{i,x} A \tau_{j,x} \rangle + \langle \tau_{i,y} C \tau_{j,y} \rangle & \langle \tau_{i,x} B \tau_{j,y} \rangle + \langle \tau_{i,y} C \tau_{j,x} \rangle \\ \langle \tau_{i,y} B \tau_{j,x} \rangle + \langle \tau_{i,x} C \tau_{j,y} \rangle & \langle \tau_{i,y} A \tau_{j,y} \rangle + \langle \tau_{i,x} C \tau_{j,x} \rangle \end{bmatrix}$$

$$\underline{\underline{\epsilon}} \begin{pmatrix} \sigma_i \\ 0 \\ 0 \end{pmatrix} : \underline{\underline{\sigma}} \begin{pmatrix} \tau_j \\ 0 \\ 0 \end{pmatrix}$$

LOCAL ELASTICITY MATRIX

COMPUTING A LOCAL ELASTICITY MATRIX



	$\tau_{i,x}$	$\tau_{i,y}$
1	-1	0
2	1	-1
3	0	1

$$\left[\begin{array}{c|c} A \langle \tau_{i,x} \tau_{j,x} \rangle + C \langle \tau_{i,y} \tau_{j,y} \rangle & B \langle \tau_{i,x} \tau_{j,y} \rangle + C \langle \tau_{i,y} \tau_{j,x} \rangle \\ \hline B \langle \tau_{i,y} \tau_{j,x} \rangle + C \langle \tau_{i,x} \tau_{j,y} \rangle & A \langle \tau_{i,y} \tau_{j,y} \rangle + C \langle \tau_{i,x} \tau_{j,x} \rangle \end{array} \right]$$

$$A_{\equiv ij}^e \quad \begin{array}{l} A_{xxij} = A \\ A_{yyij} = C \\ A_{xyij} = 0 \\ A_{yxij} = 0 \end{array}$$

$$A_{\equiv ij}^e =$$

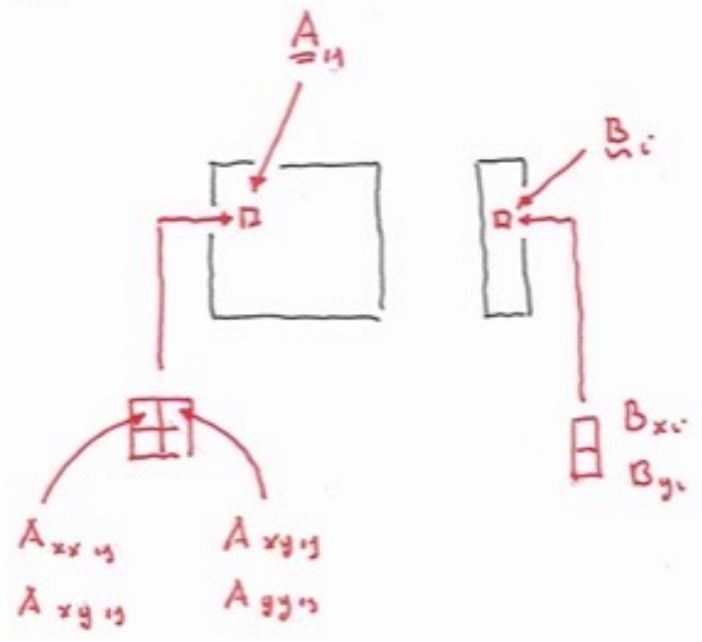
$$\left[\begin{array}{cc|cc} A & 0 & -A & B \\ 0 & C & C & -C \\ \hline A+C & -B-C & -C & B \\ -B-C & A+C & C & -A \\ \hline & & C & 0 \\ & & 0 & A \end{array} \right] \frac{1}{2}$$

$$\underline{B}_i = \ll g \cdot \hat{u}^h \gg_N$$

$\left[\begin{matrix} \tau_i \\ 0 \end{matrix} \right]$ OR $\left[\begin{matrix} 0 \\ \tau_i \end{matrix} \right]$

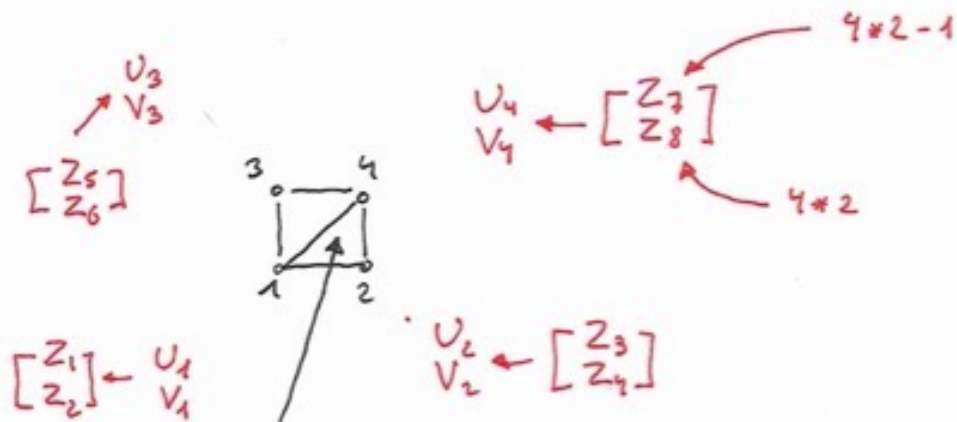
$$\left[\begin{matrix} g_x \\ g_y \end{matrix} \right]$$

$$\left[\begin{matrix} \ll g_x \tau_i \gg \\ \ll g_y \tau_i \gg \end{matrix} \right]$$



AND
 HYPER
 VECTOR !

ASSEMBLING PROCEDURE



$$A_{ij}^e = [\quad]$$

6x6 MATRIX

$$\left[\begin{array}{c} \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \end{array} \right] \left[\begin{array}{c} Z_1 \\ \vdots \\ \vdots \\ Z_8 \end{array} \right] = \left[\begin{array}{c} \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \\ \vdots \end{array} \right]$$

GLOBAL
HORIZONTAL
INDEX

$$= 2 * \underbrace{\text{GLOBAL SCALAR INDEX}}_{\text{GLOBAL}} - 1$$